COPERNICUS POD SERVICE model updates of copernicus sentinel-1,-2,-3 orbits

ABSTRACT

The European Earth observation satellite missions, Copernicus Sentinel-1, -2, and -3, require consistent orbit time series for highest-level final products. The six satellites of the three missions have been launched beginning with Sentinel-1A in April 2014 and the last launch of Sentinel-3B has been in April last year. Delivering consistent GPS-derived orbit time series for all six satellites with the different mission lengths and also following up model updates and improvements is a challenge to be fulfilled in an operational environment. Recently, several model EIGEN.GRGS.RL04.v1, the atmosphere gravity product GFZ AOD1B RL06, plus the use of single-receiver ambiguity resolution have been intensively validated based on orbit comparisons and SLR residuals, when available. Long-term analysis from offline reprocessed orbit solutions based on the updated models support the validation activities necessary for the continuous improvements of the orbital products. The current status of the model updates and the corresponding validation activities are presented for all six satellites focussing on Sentinel-1 SAR (Synthetic Aperture Radar) and Sentinel-3 altimetry missions, because they have the most demanding accuracy requirements.

INTRODUCTION – THE COPERNICUS POD SERVICE

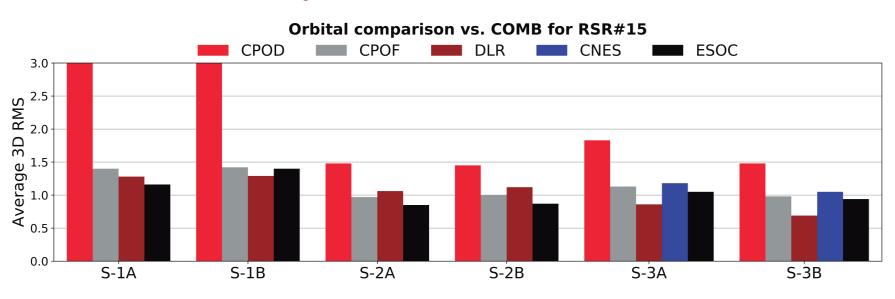
The **Copernicus POD Service (CPOD)**, part of the Copernicus PDGS and located in Tres Cantos (Madrid), is responsible for the generation of precise orbital products and auxiliary data for their use as parts of the processing chains of the PDGS.

Sentinel-3 Sentinel-1 Sentinel-2

POD CONFIGURATION AND MODELING

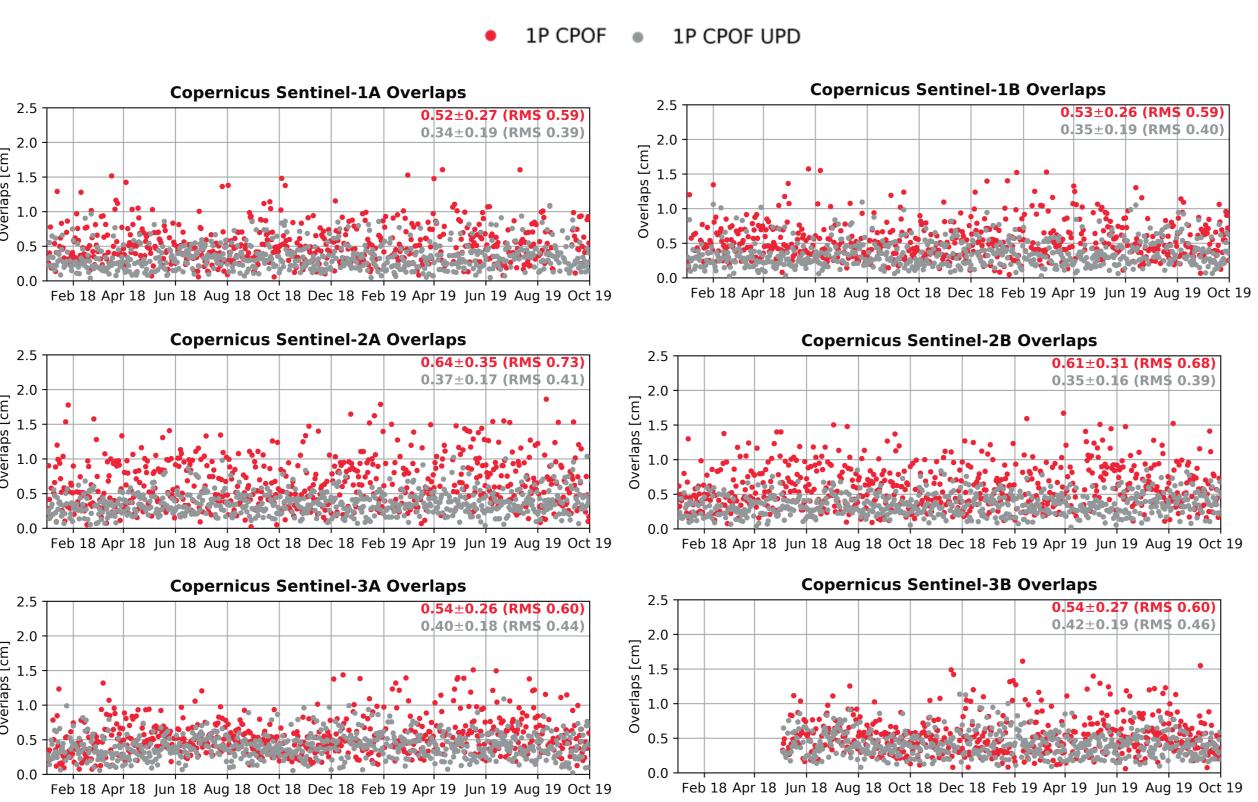
Parameter	Element	Value		
Software		NAPEOS		
Arc length		32 hours		
Reference systems	Polar motion and UT1	IERS C04 08		
	Pole model	IERS 2010 Conventions		
	Precession/Nutation	IERS 2010 Conventions		
	Radiation Pressure model	box-wing model with re-radiation		
	Earth radiation	albedo and infra-red applied		
Surfaces forces	Radiation pressure	1 per arc		
and empiricals	coefficient			
	Drag coefficients	11 per day		
	1/rev empirical	2 sets per day (along/cross-track, sine/cosine)		
GPS measurements	Relativity	IERS 2010		
	Sampling	11 sec		
	Observations	iono-free of phase and pseudo-range		
	Weight	0.8 m (pesudo-range) / 10 mm (carrier-phase)		
	Elevation angle cut-off	7 degrees		
Geometrical properties	Sentinel-1	GMV-GMESPOD-TN-0025		
	Sentinel-2	GMV-GMESPOD-TN-0026		
	Sentinel-3	GMV-GMESPOD-TN-0027		
GPS ANTEX	Filename	sen08_2025_S1_mod.atx		
		Solid Earth, Ocean, Atmospheric, Earth Pole, Ocean		
Other factors		Pole tides + Third bodies (Sun, Moon, Planets		
		DE405)		
Scenario	Element	Value		
OPERATIONS	Gravity field (static)	EIGEN.GRGS.RL03.v2 (120x120) time-dep (50x50)		
	Atmospheric gravity	AGRA (20x20) (Ray-Ponte 2003)		
	Atmospheric model	MSISE90 (Hedin, 1991)		
NEW MODELS	Gravity field (static)	EIGEN.GRGS.RL04 (120x120) time-dep (50x50		
	Atmospheric gravity	GFZ AOD L1B		
	Atmospheric model	MSISE00		

MOTIVATION: QWG RSR#15



ORBITS OVERLAPS

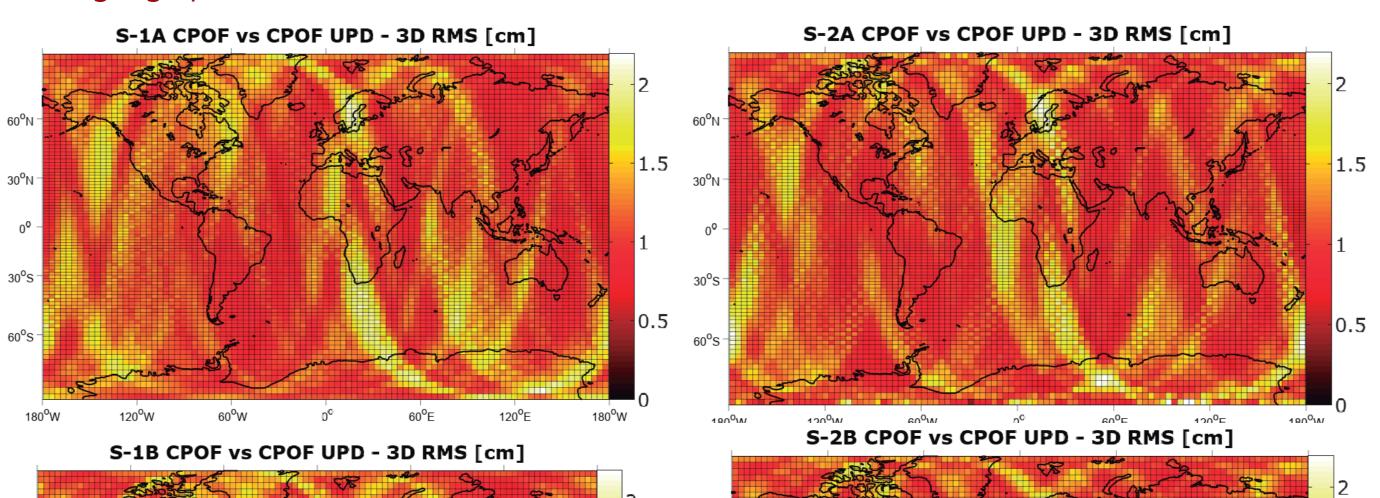
New models yield better overlaps between orbits

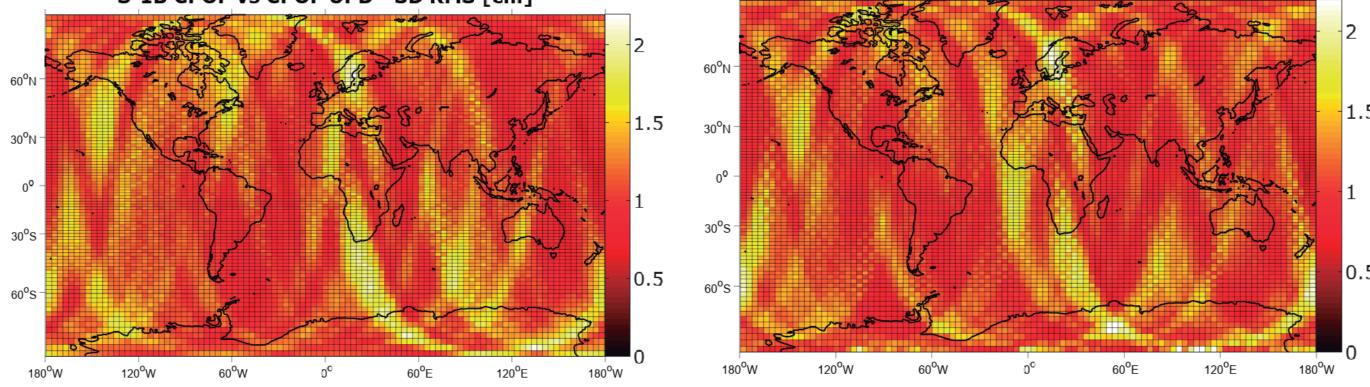


Operational CPOD solution is getting obsolete with respect to the current ate-of-the-art levels

GEOGRAPHICAL DIFFERENCES

Similar geographical 3D RMS differences for each mission account for the net contribution of the updated models





VALIDATION AGAINST BASELINE SOLUTION

New models yield systematic improvements when comparing against the baseline solution (Fernandez et al. 2019)

Comparisons against baseline solution – 3D RMS [cm]							
	SEN-1A	SEN-1B	SEN-2A	SEN2B	SEN-3A	SEN-3B	
CPOF	$\textbf{1.53} \pm \textbf{0.20}$	$\textbf{1.55} \pm \textbf{0.21}$	$\textbf{1.34} \pm \textbf{0.18}$	$\textbf{1.37} \pm \textbf{0.17}$	$\textbf{1.40} \pm \textbf{0.19}$	$\textbf{1.30} \pm \textbf{0.21}$	
CPOF UPD	$\textbf{1.41} \pm \textbf{0.22}$	$\textbf{1.44} \pm \textbf{0.25}$	$\textbf{1.02} \pm \textbf{0.11}$	$\textbf{1.07} \pm \textbf{0.12}$	$\textbf{1.09} \pm \textbf{0.10}$	$\textbf{0.95} \pm \textbf{0.09}$	

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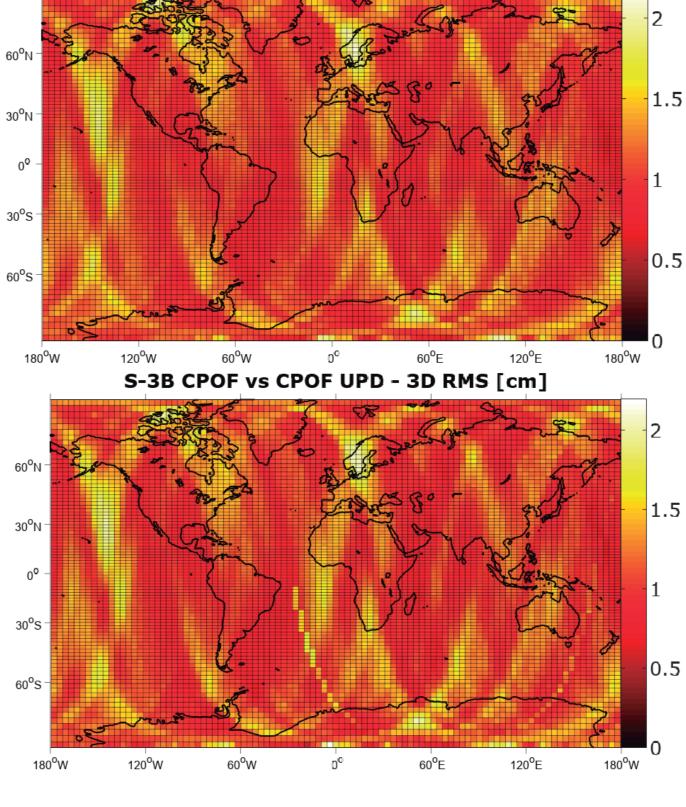
Heike Peter (2) Emilio J. Calero (1) Javier Berzosa (1) Jaime Fernández(1)

References

GMV-GMESPOD-TN-0025 Sentinel-1 properties for GPS POD: https://sentinels.copernicus.eu/web/sentinel/missions/sentinel-1/ground-segment/pod/documentation GMV-GMESPOD-TN-0026 Sentinel-2 properties for GPS POD: https://sentinels.copernicus.eu/web/sentinel/missions/sentinel-2/ground-segment/pod/documentation GMV-GMESPOD-TN-0027 Sentinel-3 properties for GPS POD: https://sentinels.copernicus.eu/web/sentinel/missions/sentinel-3/ground-segment/pod/documentation Fernández J., Peter H., Calero E.J., Berzosa J., Gallardo L.J., Féménias P., 2019. Sentinel-3A – Validation of Orbit Products at the Copernicus POD Service Hedin, Alan E. "Extension of the MSIS thermosphere model into the middle and lower atmosphere." Journal of Geophysical Research: Space Physics 96, no. A2 (1991): 1159-1172 Peter H., Calero E. J., Fernández J., Féménias P. "Copernicus POD Service - Model Updates and Validation of Sentinel-3 Orbit Determination" OSTST 2019, Chicago, IL

Acknowledgements

SENTINEL-3 SLR VALIDATION



S-3A CPOF vs CPOF UPD - 3D RMS [cm]

*Bias per SLR station

Yarragadee	5.96 mm
Greenbelt	1.56 mm
Haleakala	12.50 mm
Hartebeesthoek	9.14 mm
Graz	11.08 mm
Herstmonceux	3.99 mm
Potsdam	-7.43 mm
Matera	-3.29 mm
Wettzell	-17.03 mm

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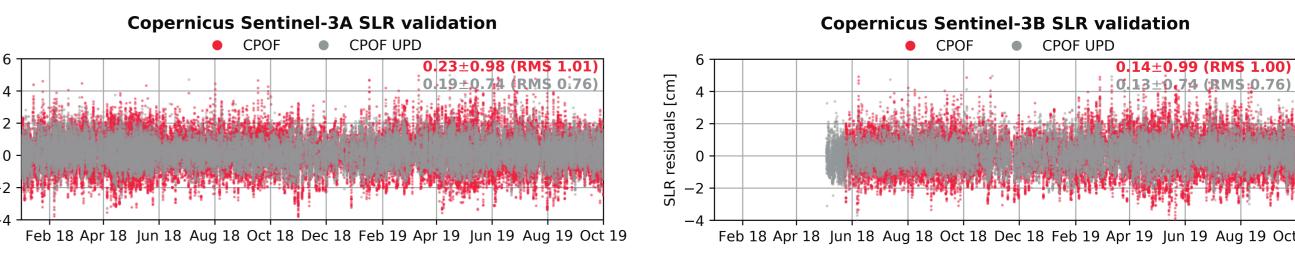
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New models yield lower dispersion of SLR residuals after removing a bias per SLR station*



CONCLUSIONS

Some effort has been devoted to update and validate different physical models at the Copernicus POD Service:

- The gravity field, atmospheric gravity and atmospheric density models have been updated.
- The assessment of overlaps as well as SLR residuals shows an improvement in accuracy for the new solutions. Similar improved figures are observed when comparing against the baseline solutions.
- The updated models yield consistent geographical differences within the Sentinel missions.









