

# Copernicus Sentinel-3 Surface Topography Mission - Cyclic Performance Report SRAL

S3A Cycle No. 80

Start date: 18/12/2021

End date: 13/01/2022

S3B

Cycle No. 61

Start date: 28/12/2021

End date: 23/01/2022

Reference: S3MPC-STM\_CPR\_0005-080-061

Issue 1.0 - 01/02/2022

Contract: 4000136824/21/I-BG

Open/Public/Público

#### Disclaimer



#### **CHRONOLOGY ISSUES**

Issue	Date	Object	Written by	Checked by	Approved by
1.0	27/01/2022	Initial issue	P. García	J. Aublanc	G. Jettou

### **ACCEPTANCE**

	CLIENT: ESA	SUPPLIER: CLS	
Name	Pierre Féménias	Ghita Jettou	
Function	ESA Technical Officer	MPC Service Manager	

### **LIST OF CONTENTS**

1	Intro	oduction	. 6
2	SRA	L performance overview	. 7
	2.1	SRAL Cycle Performance	. 7
	2.2	SRAL Mission Performance	. 7
	2.3	SRAL anomalies and events	. 7
3	Proc	cessing baseline	. 7
4	SRA	L monitoring report	. 7
	4.1	Cycle SRAL Performance	8
	4.1.	1 CAL1 LRM	. 9
	4.1.	2 CAL1 SAR	15
	4.1.	3 CAL2	23
	4.1.	4 AutoCal	27
	4.1.	5 Thermal behaviour	30
	4.1.	6 Cyclic Status Summary	31
	4.2	Mission Status Summary	32
5	Calil	bration with Transponder	45



### **LIST OF TABLES**

Table 1: Processing baseline and IPF detailsError! Book	mark not defined.
Table 5-1. Results of Crete TRP passes processing	46
Table 5-2. Results of Svalbard TRP passes processing	46
LIST OF FIGURES	
Figure 1: S3A and S3B cycles chronology	6
Figure 4-1. Location of the S3A CAL1 LRM measurements	9
Figure 4-2. Location of the S3B CAL1 LRM measurements.	9
Figure 4-3. S3A CAL1 LRM Ku Time Delay trend	10
Figure 4-4. S3B CAL1 LRM Ku Time Delay trend	10
Figure 4-5. S3A CAL1 LRM Ku Power Trend	11
Figure 4-6. S3B CAL1 LRM Ku Power Trend	11
Figure 4-7. S3A CAL1 LRM Ku PTR width trend	12
Figure 4-8. S3B CAL1 LRM Ku PTR width trend	12
Figure 4-9. S3A CAL1 LRM Ku PTR secondary lobes Power and Position within the	PTR waveform.13
Figure 4-10. S3B CAL1 LRM Ku PTR secondary lobes Power and Position within t	he PTR waveform. 13
Figure 4-11. S3A CAL1 LRM PTR secondary lobes characterisation. The inter-ann and standard deviation (in dBx10 $^{-2}$ ) of each of the secondary lobes are shown	, , ,
Figure 4-12. S3B CAL1 LRM PTR secondary lobes characterisation. The inter-ann and standard deviation (in $dBx10^{-2}$ ) of each of the secondary lobes are shown	, , , ,
Figure 4-13. S3A Location of the CAL1 SAR measurements	15
Figure 4-14. S3B Location of the CAL1 SAR measurements	15
Figure 4-15. S3A CAL1 SAR Ku Time Delay trend	16
Figure 4-16. S3B CAL1 SAR Ku Time Delay trend	16
Figure 4-17. S3A CAL1 SAR Ku Power Trend	17
Figure 4-18. S3B CAL1 SAR Ku Power Trend	17
Figure 4-19. S3A CAL1 SAR Ku PTR width trend	18
Figure 4-20. S3B CAL1 SAR Ku PTR width trend	18
Figure 4-21. S3A CAL1 SAR Ku PTR secondary lobes Power and Position within the	e PTR waveform.19
Figure 4-22. S3B CAL1 SAR Ku PTR secondary lobes Power and Position within the	ne PTR waveform.19
Figure 4-23. S3A CAL1 SAR Ku PTR secondary lobes characterisation. The interad dB/year) and standard deviation (in dBx10- $^2$ ) of each of the secondary lobes	
Figure 4-24. S3B CAL1 SAR Ku PTR secondary lobes characterisation. The inter-a dB/year) and standard deviation (in dBx10-2) of each of the secondary lobes	
Figure 4-25. S3A CAL1 SAR Ku Phase & Power intra-burst corrections slopes over	r the cycle21
Figure 4-26. S3B CAL1 SAR Ku Phase & Power intra-burst corrections slopes ove	r the cycle21



Figure 4-27. S3A Pulse by pulse standard deviations of the CAL1 SAR Ku Power and corrections	
Figure 4-28. S3B Pulse by pulse standard deviations of the CAL1 SAR Ku Power and corrections	
Figure 4-29. S3A Location of the CAL2 SAR measurements	23
Figure 4-30. S3B Location of the CAL2 SAR measurements	23
Figure 4-31. S3A Averaged CAL2 SAR Ku and C waveforms over the cycle	24
Figure 4-32. S3B Averaged CAL2 SAR Ku and C waveforms over the cycle	24
Figure 4-33. S3A CAL2 SAR Ku waveforms right (blue) and left (red) sides Slope over	the period.25
Figure 4-34. S3B CAL2 SAR Ku waveforms right (blue) and left (red) sides Slope over	the period.25
Figure 4-35. S3A CAL2 SAR Ku waveforms right (blue) and left (red) sides Standard Devia	
Figure 4-36. S3B CAL2 SAR Ku waveforms right (blue) and left (red) sides Standard Devia	
Figure 4-37. S3A Location of the AutoCal measurements.	27
Figure 4-38. S3B Location of the AutoCal measurements	
Figure 4-39. S3A AutoCal measurements: Corrected - Reference. Averaged over the	
Figure 4-40. S3B AutoCal measurements: Corrected - Reference. Averaged over the	cycle28
Figure 4-41. S3A Ku band Autocal Cycle Progression	29
Figure 4-42. S3B Ku band Autocal Cycle Progression.	29
Figure 4-43. S3A Selected group of Thermistors time series on CAL1 SAR mode. The tem averaged for each calibration product.	•
Figure 4-44. S3B Selected group of Thermistors time series on CAL1 SAR mode. The tem averaged for each calibration product.	•
Figure 4-45. S3A CAL1 SAR Ku Time Delay Whole Mission Trend	33
Figure 4-46. S3B CAL1 SAR Ku Time Delay Whole Mission Trend	33
Figure 4-47. S3A CAL1 SAR Ku Power Whole Mission Trend	34
Figure 4-48. S3B CAL1 SAR Ku Power Whole Mission Trend	34
Figure 4-49. S3A CAL1 SAR Ku Power series and long term trend extrapolation with $\mu$	ower model.35
Figure 4-50. S3B CAL1 SAR Ku Power series and long term trend extrapolation with p	oower model.35
Figure 4-51. S3A CAL1 SAR Ku PTR Width Whole Mission Trend	36
Figure 4-52. S3B CAL1 SAR Ku PTR Width Whole Mission Trend	36
Figure 4-53. S3A CAL1 SAR Ku Phase & Power intra-burst corrections slopes along the	
Figure 4-54. S3B CAL1 SAR Ku Phase & Power intra-burst corrections slopes along the	he whole mission.
Figure 4-55. S3A Slope at each side of the CAL2 Ku waveform, averaged over the wh	nole mission.38
Figure 4-56. S3B Slope at each side of the CAL2 Ku waveform, averaged over the wh	
Figure 4-57. S3A Autocal measurements: Corrected - Reference. Averaged over the v	vhole mission. 39
Figure 4-58, S3B Autocal measurements: Corrected - Reference, Averaged over the v	whole mission, 39



Figure 4-59. S3A AutoCAL Ku attenuation whole mission progression. Difference in dB with respect the first attenuation value, for each attenuation step40	
Figure 4-60. S3B AutoCAL Ku attenuation whole mission progression. Difference in dB with respect the first attenuation value, for each attenuation step40	to
Figure 4-61. S3A USO frequency impact in the range	
Figure 4-62. S3B USO frequency impact in the range41	
Figure 4-63. S3A Selected CAL1 SAR thermistors series along the whole mission42	
Figure 4-64. S3B Selected CAL1 SAR thermistors series along the whole mission42	
Figure 5-1. Range Bias Results47	
Figure 5-2. Datation Bias Results	
Figure 5-3. Alignment Results49	
Figure 5-4. Stack Noise Results50	



# 1 Introduction

The purpose of this document is to report on the performance and data quality of the Copernicus Sentinel-3 Surface Topography Mission (STM) LAND products. The constellation currently includes Sentinel-3A and Sentinel-3B altimetry satellites. This document is associated with data dissemination on a cyclic basis and is generated a few days after the end of Sentinel-3B cycle.

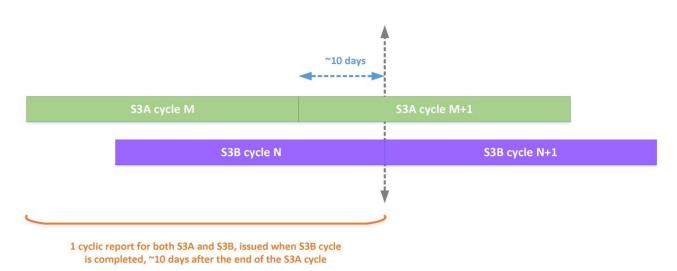


Figure 1: S3A and S3B cycles chronology

The CAL Level 1B products assessed hereafter are produced by the ESA Sentinel-3 LAND Processing Centre. One of the main goals of the cyclic report is to detect and report as quickly as possible any events, or anomaly, impacting the data quality. Subsequently, the assessments are made on the Short Time Critical (STC) products, generally delivered 48 hours after data acquisitions. Differences are expected with the Non Time Critical (NTC) products, for which the orbit data and several geophysical corrections are consolidated.

The main objectives of this document are:

- To provide a data quality assessment of the Sentinel-3 CAL Level 1B STC products
- ➤ To report on any changes likely to impact data quality at any level, from instrument status to software configuration.
- To present the major useful results for S3A cycle 80, from 18/12/2021to 13/01/2022.
- > To present the major useful results for S3B cycle 61, from 28/12/2021 to 23/01/2022.



# 2 SRAL performance overview

### 2.1 SRAL Cycle Performance

The SRAL behaviour in this cycle is nominal for the two Sentinel-3 missions. The absolute values and the local trends are as expected for the assessed calibration parameters. These parameters are related to the calibration modes CAL1, CAL2, AutoCal, and from the USO files and thermal data. From what we observe in the figures and the cycle status summary in 4.1, the performances are in line with previous cycles.

### 2.2 SRAL Mission Performance

The long-term behaviour of the SRAL instruments of both Sentinel-3 missions is nominal, compliant with the missions' requirements. No warning is raised currently. The power decay at the beginning of the S3A mission (-1 dB/year) is now under control (-0.3 dB/year) after a slow stabilization. A detailed summary of the health of the altimeters instruments is done in 4.2, together with explanations of different events along the missions.

#### 2.3 SRAL anomalies and events

No SRAL anomalies or events are to be reported in this cycle.

# 3 Processing baseline

Table 1 details the versions of the Processing Baseline (PB), and Level-1 and Level-2 Instrument Processing Facility software used for the products assessed. This is part of the Baseline Collection (BC) 004.

	Cycle	Processing Baseline	IPF SM2 version	IPF SR1 version	IPF MW1 version
Sentinel-3A	67	2.72	06.19	06.30	06.11
Sentinel-3B	48	1.49		06.19 06.20	

Table 1: Processing baseline and IPF details

The evolutions of the Sentinel-3 STM Processing Baseline since July 2016, end of commissioning phase, are summarized in the "Sentinel Online" Web pages:

https://sentinels.copernicus.eu/web/sentinel/technical-guides/sentinel-3-altimetry/processing-baseline

# 4 SRAL monitoring report

The SRAL instrumental calibration is assessed during the mission. Several parameters are monitored and analysed in detail in order to characterise the altimeter performance along the mission lifetime.

Two main groups of calibration parameters are monitored.

The first is derived from the Point Target Response (PTR) calibration in CAL1 mode. The PTR signal follows the same circuitry path as the science waveforms within the calibration loop. The delay caused by the



travel through the calibration path can be measured and afterwards compensated in the total range computation. The power figure of the PTR signal when traveling through the instrument also needs to be monitored and the science waveforms need to be compensated for this power level, which can drift along a mission. Moreover, there are a collection of other parameters to be checked, such as the PTR width and the secondary lobes features. These CAL1 parameters are produced separately for LRM and SAR modes, as they follow different instrumental paths, and also they are duplicated for Ku-band and C-band. Moreover there are different options for characterising the delay (maximum position or CoG position) and power (maximum power or total power) of the closed loop signal.

The second is related to the Instrument Transfer Function, measured by the CAL2 mode. The science waveforms spectra is distorted by the on-board instrumental hardware sections. Therefore, in order to retrieve the original echo shape, we need to compensate for this effect. Several parameters are derived from the analysis of the CAL2 waveforms for characterizing it and dissect any feature along the mission lifetime. The CAL2 waveform is the same for both modes LRM and SAR, but there is a distinction between bands Ku and C.

Additionally, for SAR mode, the two intra-burst corrections are monitored: they are the power and phase progressions within a burst. Science pulses within a burst are to be corrected for these expected burst variations. Some characteristics are computed for describing and following up their behaviour along the S3 missions.

The Autocal parameters monitor the actual attenuation values for each on-board ATT step. They are to be used in the L1B processing for considering the real attenuation value, instead of the commanded one on-board.

It is also of major importance the monitoring of the on-board clock (USO). The altimeter clock counter, responsible for computing the echo travel time, has a multiplicative impact in the range determination. The platform clock is responsible for the overall platform instruments datation. Their stability and performance are to be supervised along the mission.

Finally, the data coming from the thermistors located in the different sections of the on-board HW (HKTM products) is to be analysed in order to check the relation of any calibration parameters anomaly with the thermal behaviour, and find solutions for modelling the instrument characterisation (for instance orbital oscillations) if needed.

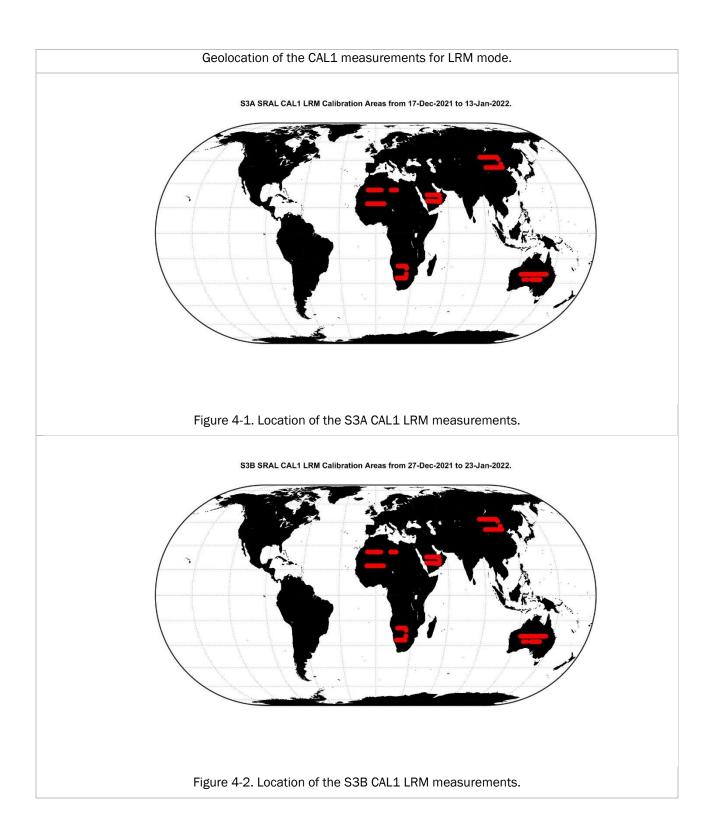
An important remark is to be made: although we can see a certain drift of a specific calibration parameter along the mission, this is not to be considered as a warning for the quality of the science data, as long as the instrumental calibration is correctly applied during the science data processing. A warning shall be raised in the scenario of a calibration parameter value approaching the mission requirement bounds.

# 4.1 Cycle SRAL Performance

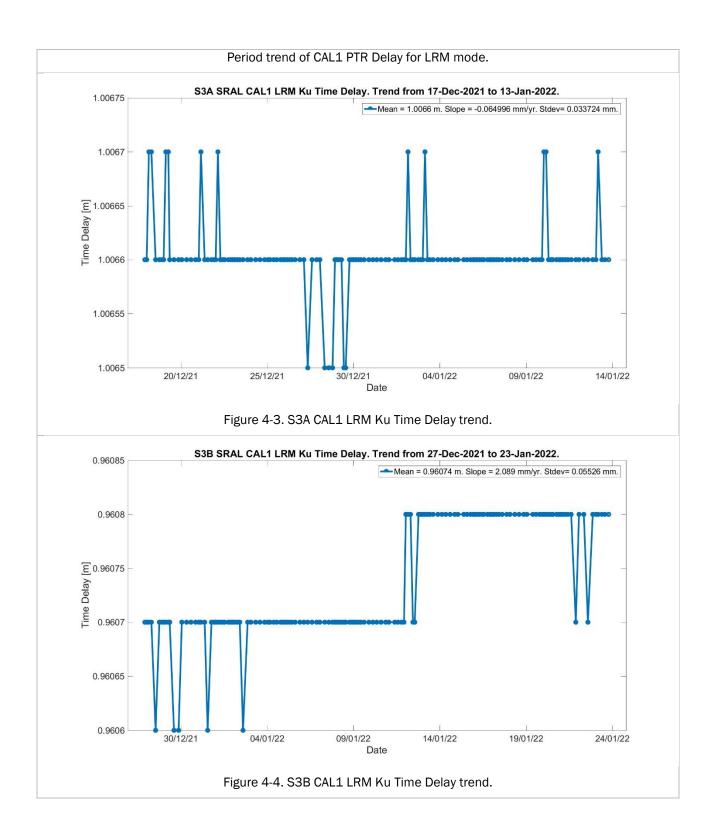
In this chapter, the monitoring of all calibration modes main parameters for the S3A and S3B missions is depicted in figures (only Ku band). A brief analysis of the cycle results is exposed at the end of this chapter.



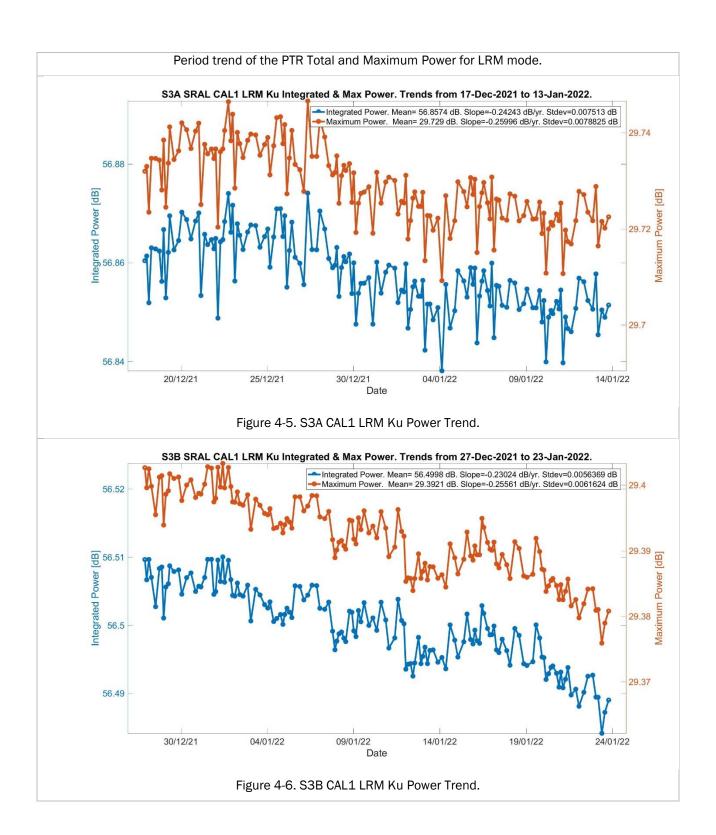
# 4.1.1 CAL1 LRM



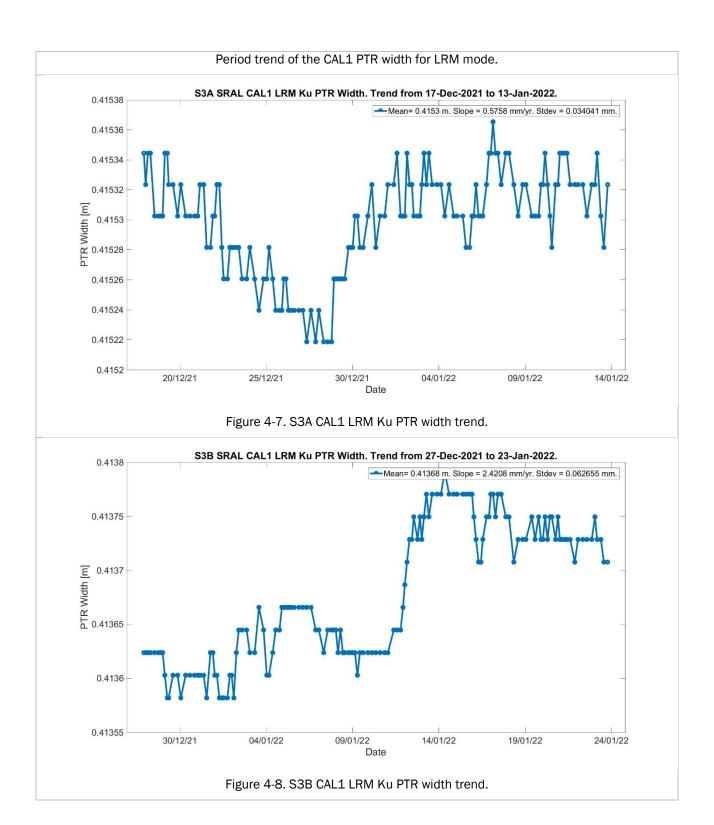




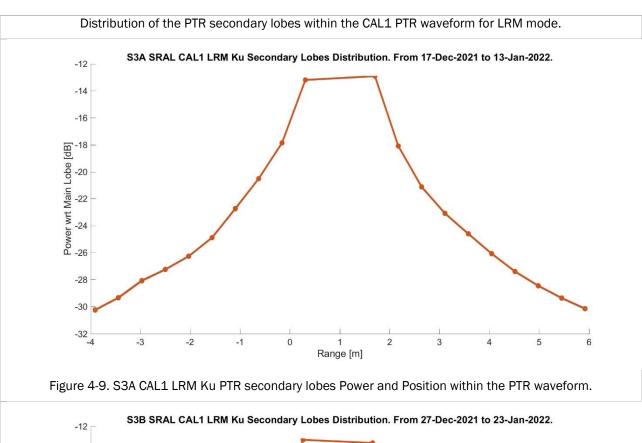


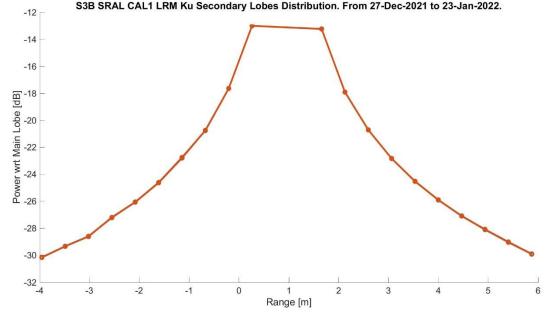


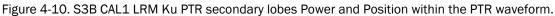
















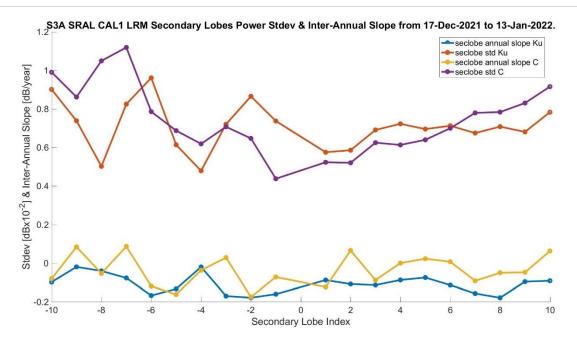


Figure 4-11. S3A CAL1 LRM PTR secondary lobes characterisation. The inter-annual slope (in dB/year) and standard deviation (in dBx10-2) of each of the secondary lobes are shown.

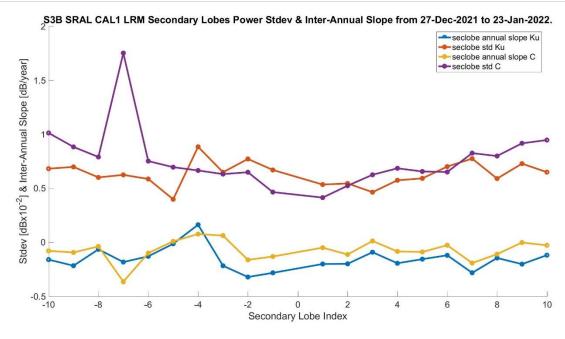
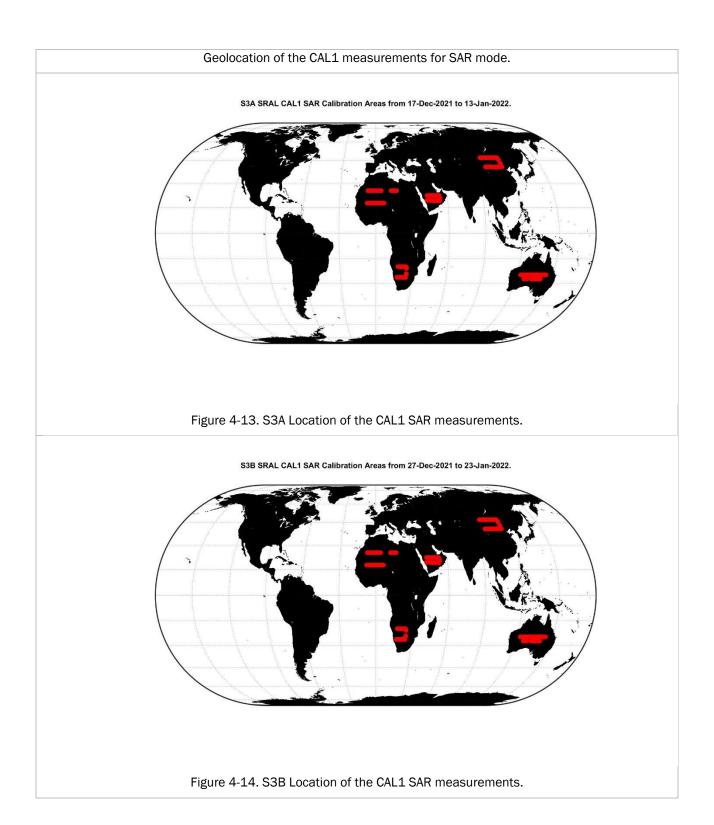


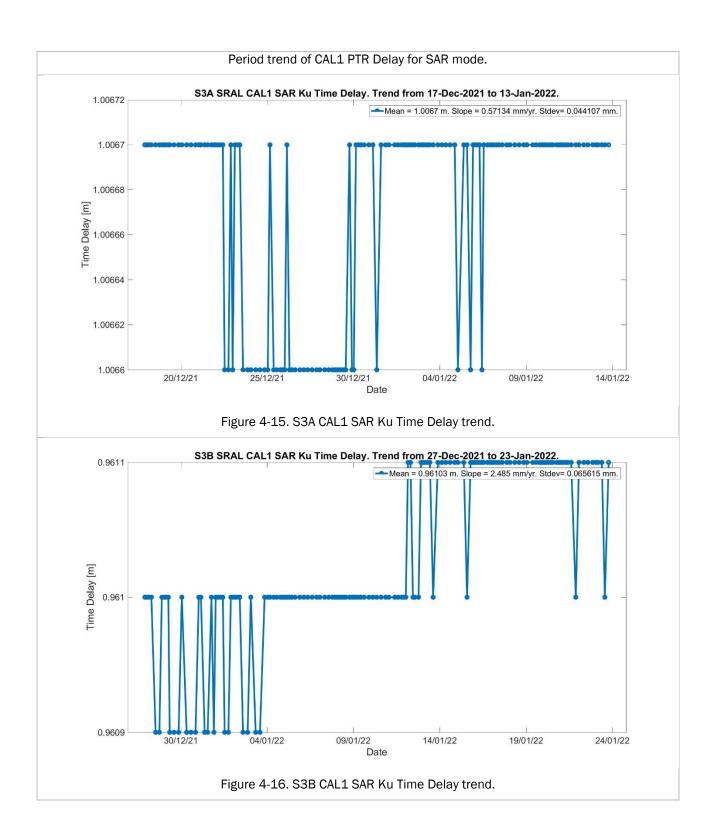
Figure 4-12. S3B CAL1 LRM PTR secondary lobes characterisation. The inter-annual slope (in dB/year) and standard deviation (in dBx10-2) of each of the secondary lobes are shown.



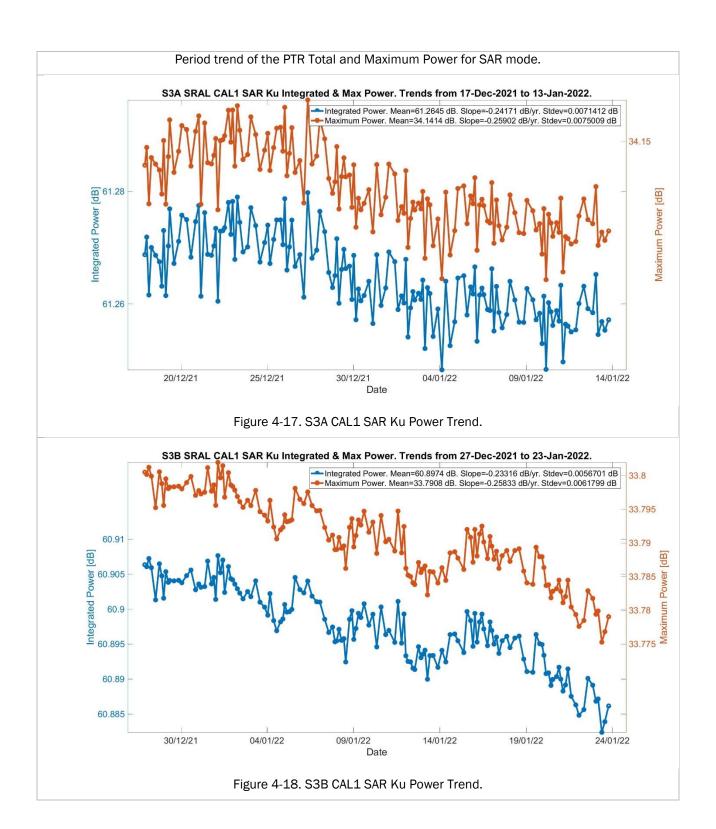
# 4.1.2 CAL1 SAR



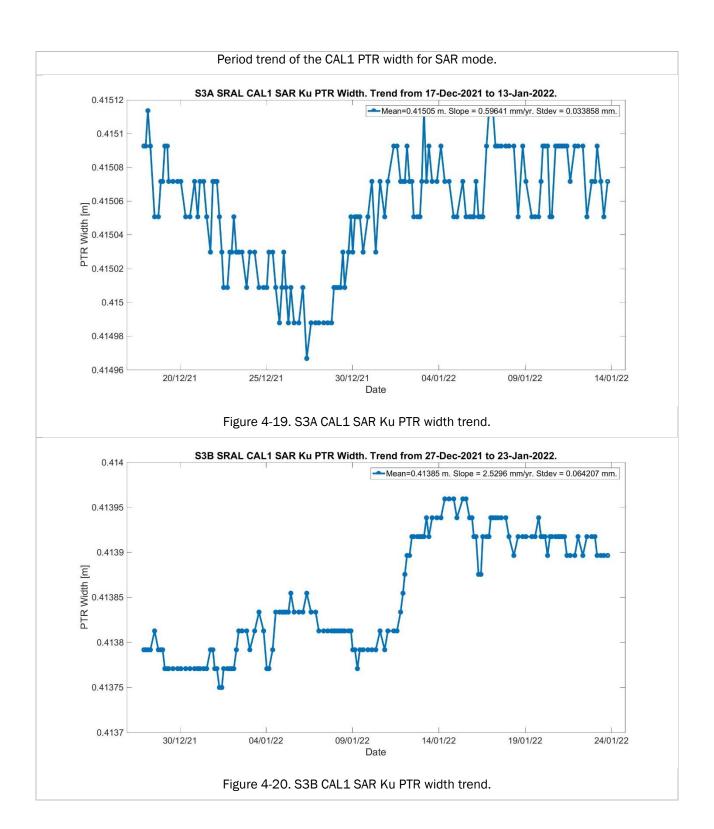














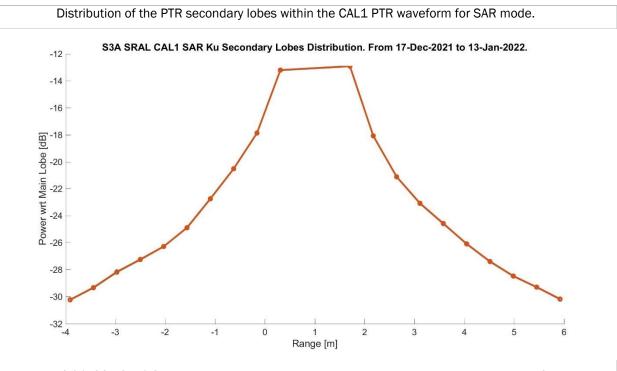


Figure 4-21. S3A CAL1 SAR Ku PTR secondary lobes Power and Position within the PTR waveform.

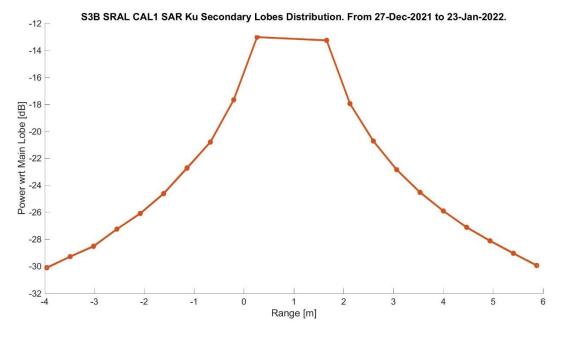


Figure 4-22. S3B CAL1 SAR Ku PTR secondary lobes Power and Position within the PTR waveform.





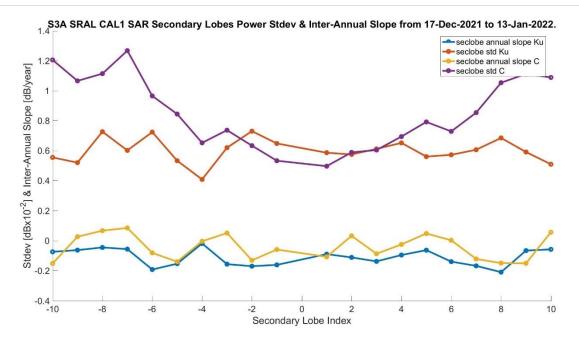


Figure 4-23. S3A CAL1 SAR Ku PTR secondary lobes characterisation. The inter-annual slope (in dB/year) and standard deviation (in dBx10<sup>-2</sup>) of each of the secondary lobes are shown.

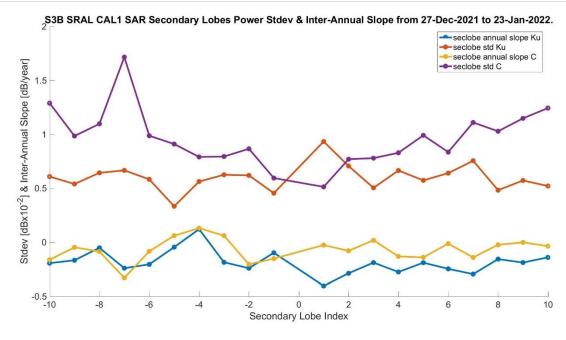


Figure 4-24. S3B CAL1 SAR Ku PTR secondary lobes characterisation. The inter-annual slope (in dB/year) and standard deviation (in dBx10<sup>-2</sup>) of each of the secondary lobes are shown.



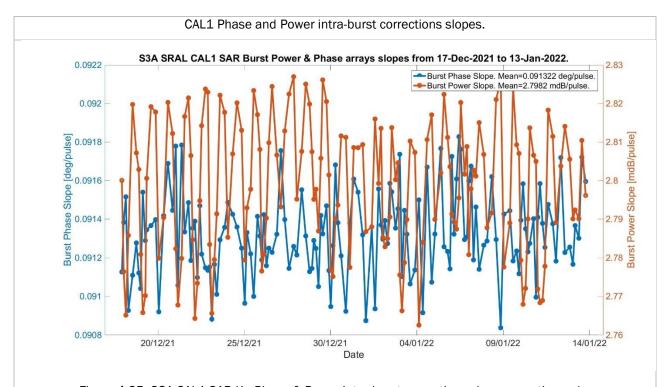


Figure 4-25. S3A CAL1 SAR Ku Phase & Power intra-burst corrections slopes over the cycle.

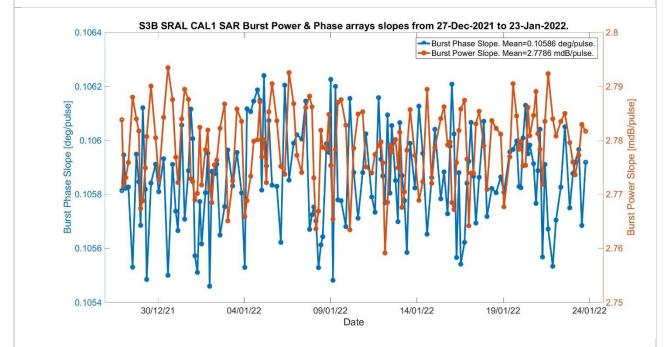


Figure 4-26. S3B CAL1 SAR Ku Phase & Power intra-burst corrections slopes over the cycle.



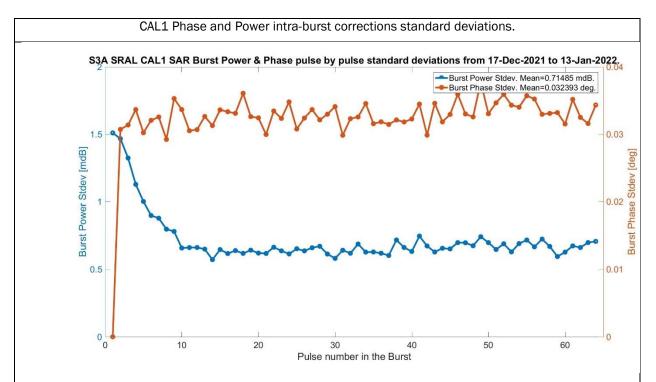


Figure 4-27. S3A Pulse by pulse standard deviations of the CAL1 SAR Ku Power and Phase intra-burst corrections.

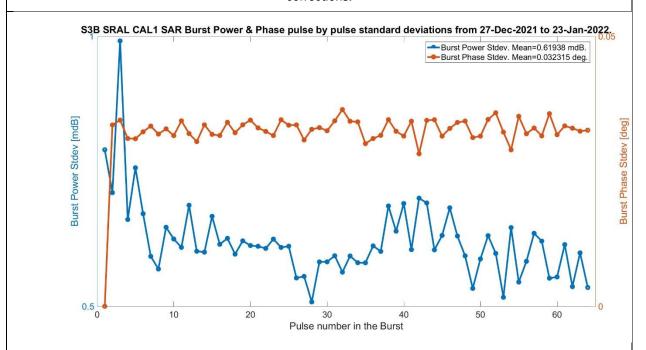
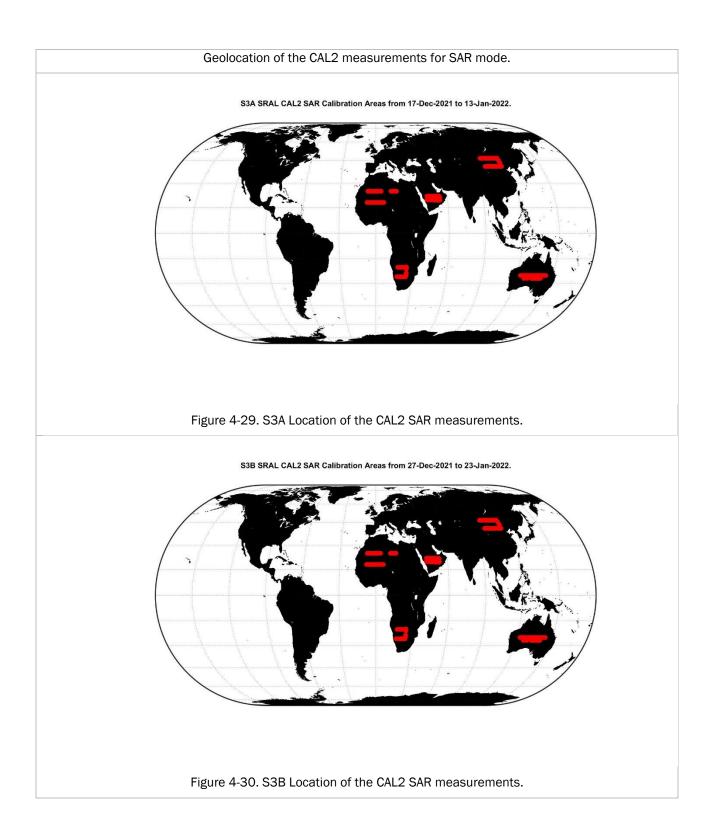


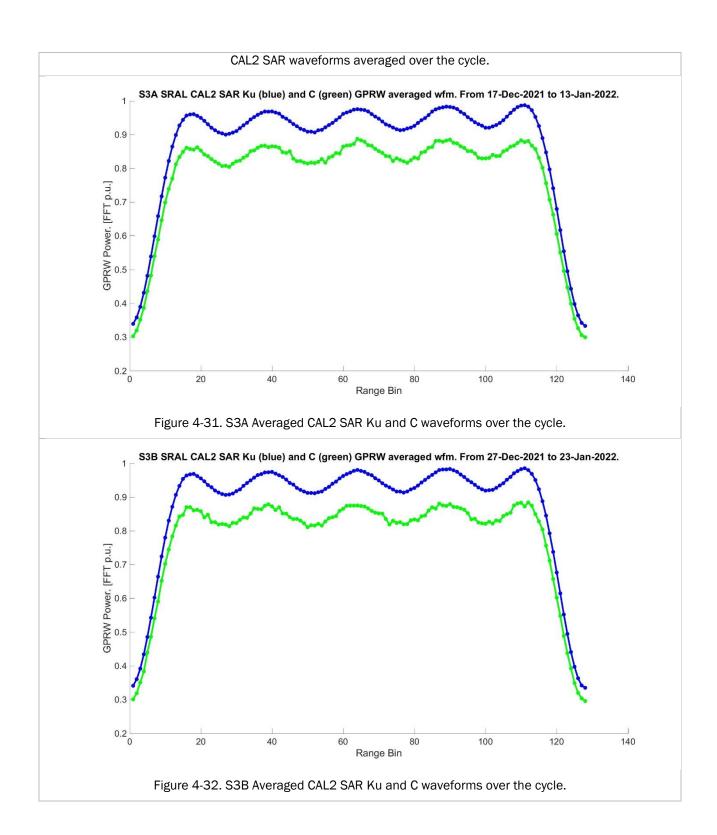
Figure 4-28. S3B Pulse by pulse standard deviations of the CAL1 SAR Ku Power and Phase intra-burst corrections.



# 4.1.3 CAL2









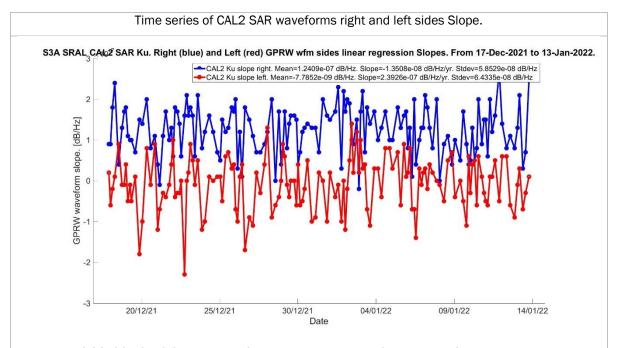
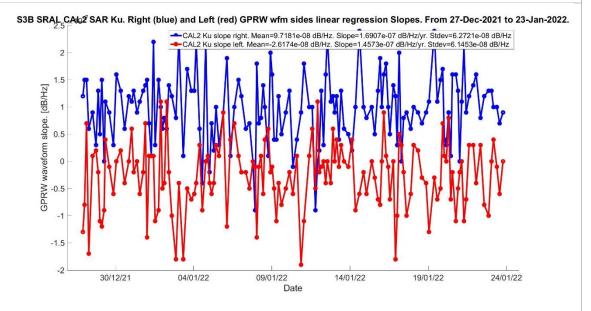
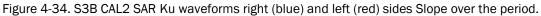


Figure 4-33. S3A CAL2 SAR Ku waveforms right (blue) and left (red) sides Slope over the period.







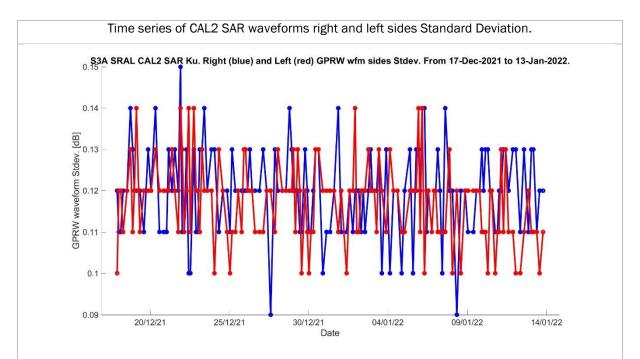


Figure 4-35. S3A CAL2 SAR Ku waveforms right (blue) and left (red) sides Standard Deviation over the period.

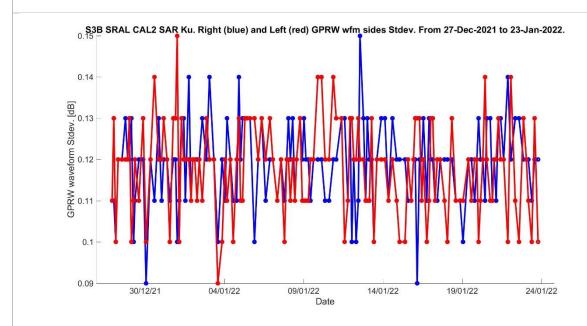
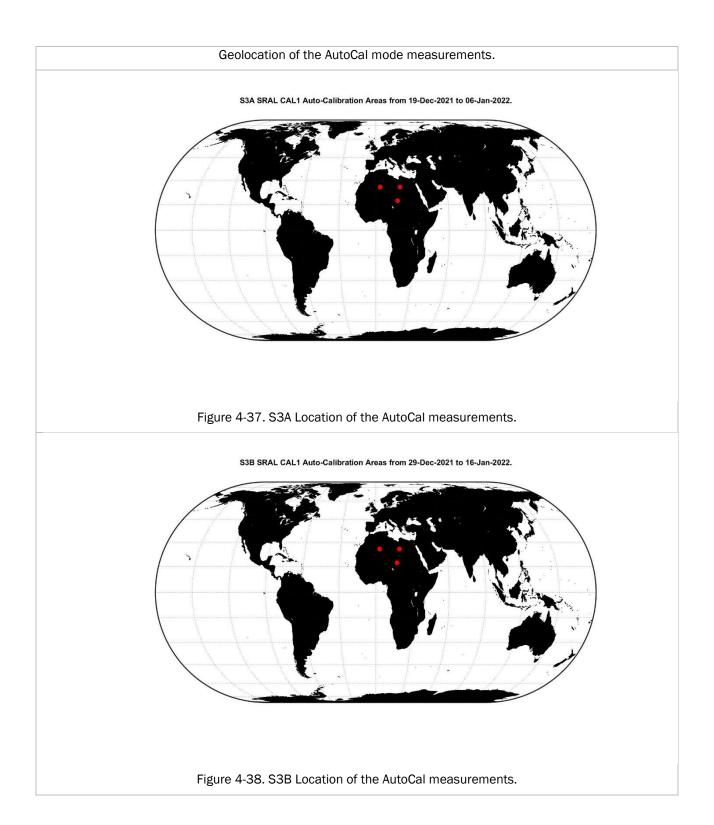


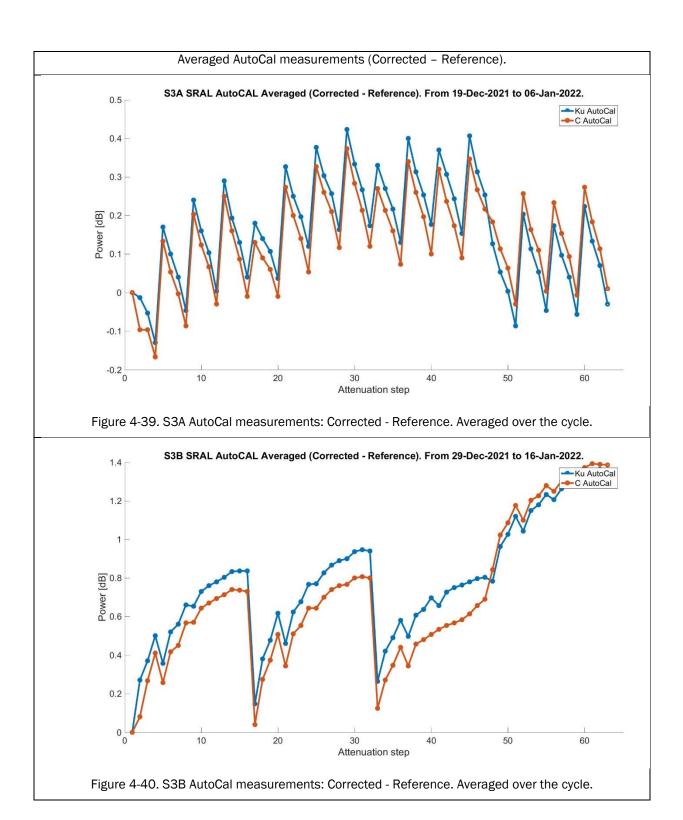
Figure 4-36. S3B CAL2 SAR Ku waveforms right (blue) and left (red) sides Standard Deviation over the period.



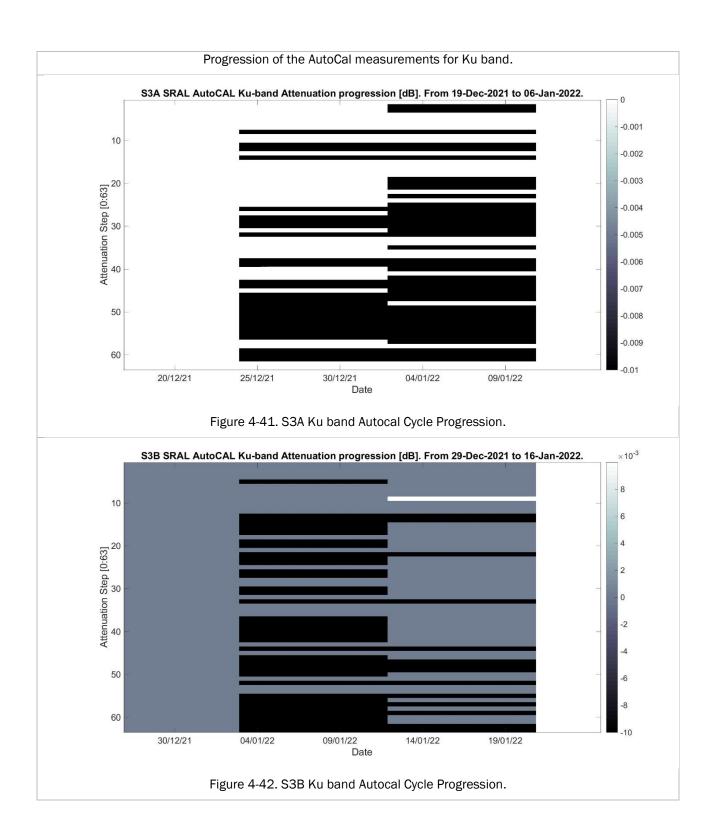
# 4.1.4 AutoCal







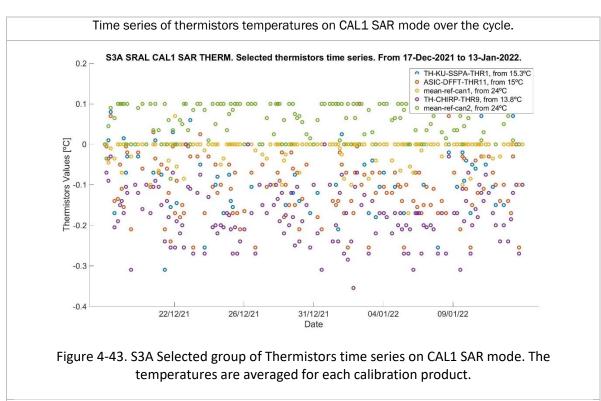






### 4.1.5 Thermal behaviour

The CAL1 SAR mode is assumed to be representative of the general SRAL thermal behaviour.



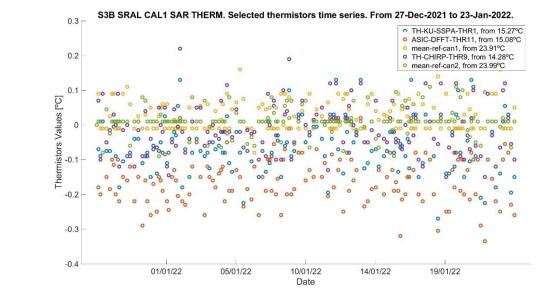


Figure 4-44. S3B Selected group of Thermistors time series on CAL1 SAR mode. The temperatures are averaged for each calibration product.



### 4.1.6 Cyclic Status Summary

This section is dedicated to a summary of the cyclic performances and status of the altimeter parameters exposed in sections 4.1.1 to 4.1.5. It covers both S3A and S3B missions.

All the S3A and S3B parameters collected show a nominal behaviour during this cycle.

In general, the LRM and SAR performances are similar for a given band (Ku or C).

The S3A CAL1 power trend for Ku band is no longer close to -1 dB/year as at the first cycles of the mission. It presents a decreasing trend but much less steep than at BOM.

All power slopes are below absolute values of 1 dB/year for both modes and bands.

The CAL1 width drifts are several orders of magnitude below the nominal PTR width value.

CAL2 parameters are stable and nominal. They are similar between the two missions.

The secondary lobes present a dissymmetry that is evolving along the mission, with notable changes after a restart of the instrument. Generally, in the figures of the characterisation of the secondary lobes during the cycle, we can see differences of standard deviation between some secondary lobes. The left-side secondary lobes tend to show higher standard deviation than the right-side ones.

AutoCal tables are nominal for both missions, and present very different attenuation steps arrays. This is not due to a fundamental difference between the S3A and S3B instruments design, but due to a different strategy for reaching the same theoretical attenuation steps values.

The thermistors values are showing generally a stable series over the analysed period.

All these observations are related to the different SRAL calibration parameters during this cycle. A whole mission monitoring is developed in section 4.2. Some of the cycle behaviours could give us the idea of a significant change of rate, but when observed within the full mission scenario, they come into scale.



### 4.2 Mission Status Summary

The main SAR Ku L1b calibration parameters series of S3A and S3B missions are gathered and plotted in this section, in order to observe their whole missions behaviour.

The selected calibration parameters are:

- CAL1 time delay
- CAL1 power
- PTR width
- Burst corrections (power and phase) and their slopes
- CAL2 waveform ripples shape, plus the waveforms slopes and de-trended standard deviations
- Autocal averaged differences and attenuation progression
- USO correction

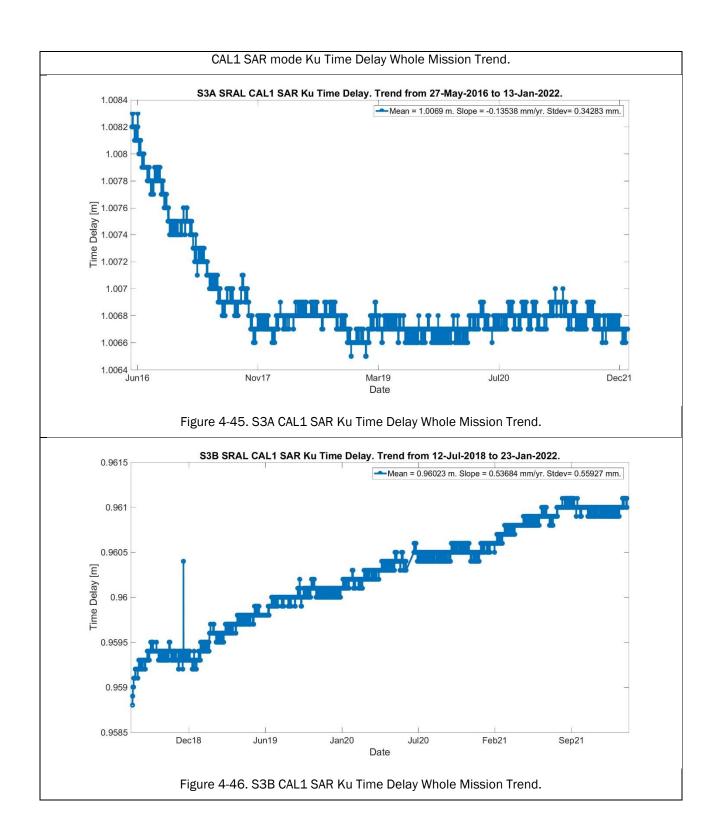
The SAR mode thermistors series are also plotted after reading the CAL ISP TM products.

Additionally, it is represented a simulation (power model fitting) of the S3A CAL1 SAR Ku Integrated Power for 20 years of mission, in order to foresee how long the SRAL Power would meet the mission requirements based in the current behaviour. This need comes from the warning raised at BOM due to high CAL1 Power trends.

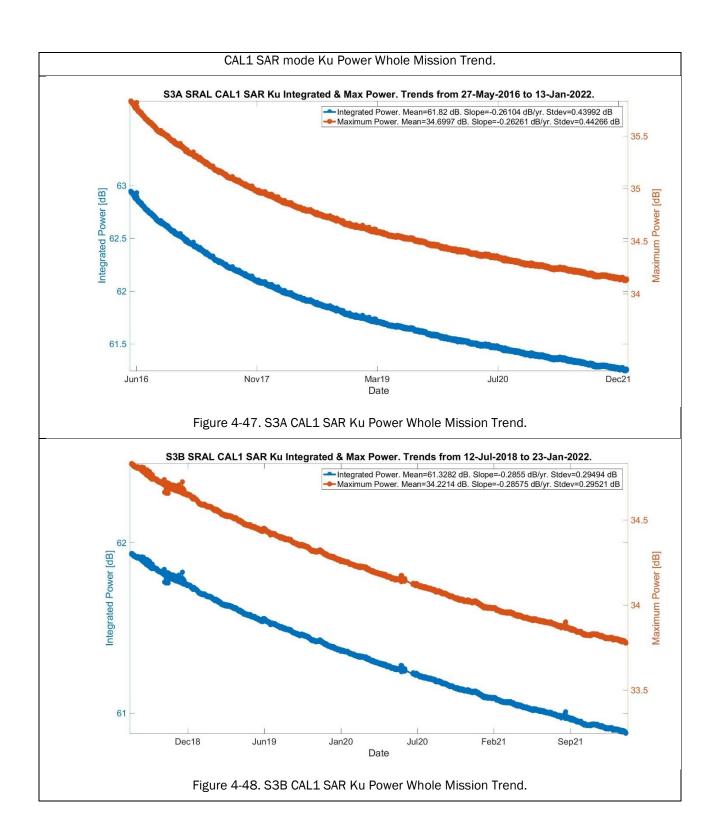
The "whole mission" figures and tables avoid a period at BOM. For the S3A mission, the considered series starts at 27/5/2016 (Cycle 4, orbit 302). For the S3B mission it starts at 12/7/2018 (Cycle 10, orbit 210). The BOM period include calibration parameters behaviours that, if included in our monitoring results, can disturb the analysis and projections made for the rest of the mission. At BOM the main parameters tend to show opposite trends with respect to the routine phase, and in the case of S3B, several operational mode changes cause jumps in the series.

The altimeter USO clock frequency has a major multiplicative impact in the determination of the altimeter range. The USO clock is the one that drives the chirp generation and controls the acquisition time (window delay or tracker range) of the returned echo signal. We depict the USO frequency impact on the altimeter range. The USO impact in the range can change around an orbit considering an elliptical orbit and the variations on the surface elevations, but these differences are far below the nominal absolute values. In addition, the temperatures on-board can make the clock suffer frequency fluctuations, but as we will see in the figures, there are no visible effects of this kind so far.











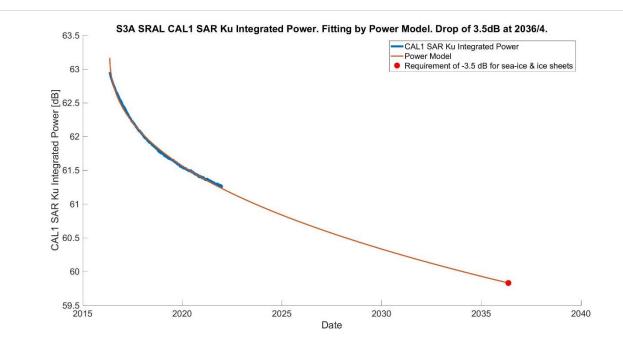
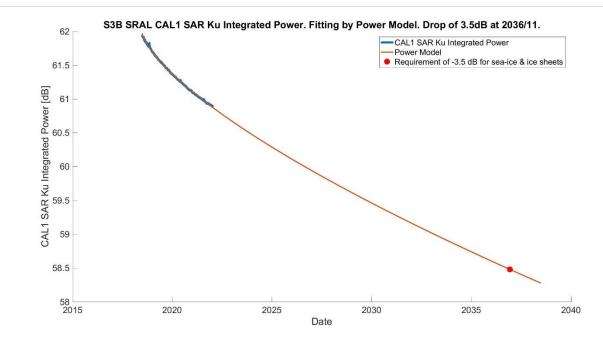
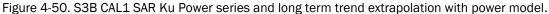
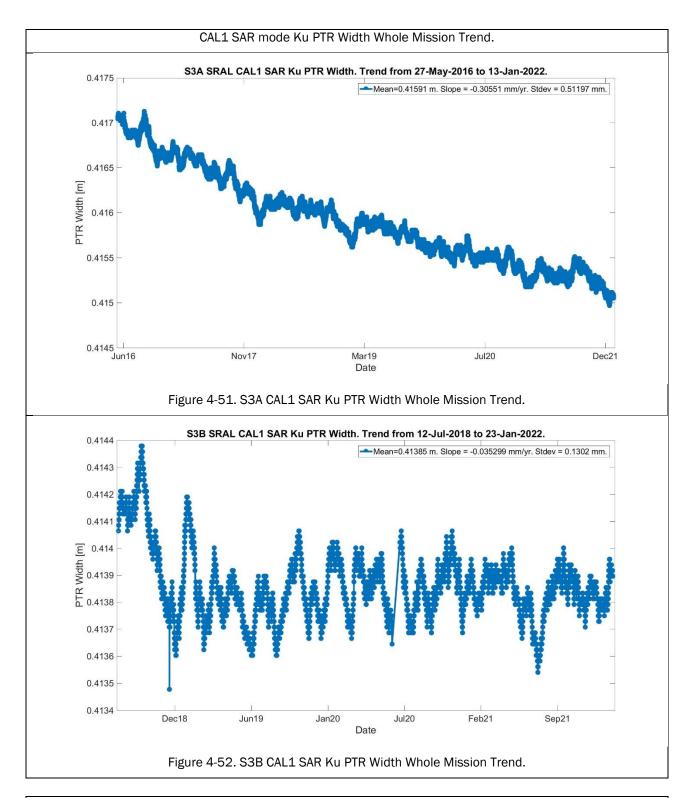


Figure 4-49. S3A CAL1 SAR Ku Power series and long term trend extrapolation with power model.









Whole Mission Trend of CAL1 SAR mode Ku Phase & Power intra-burst corrections slopes.



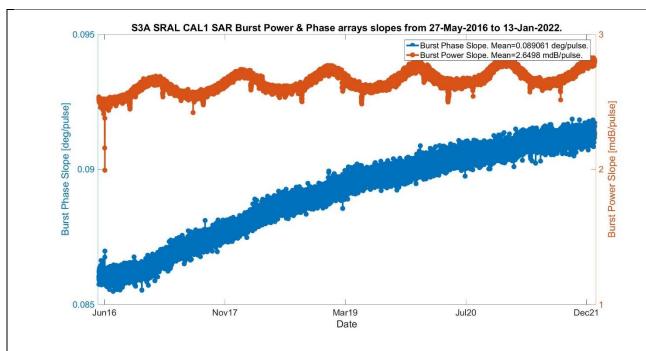


Figure 4-53. S3A CAL1 SAR Ku Phase & Power intra-burst corrections slopes along the whole mission.

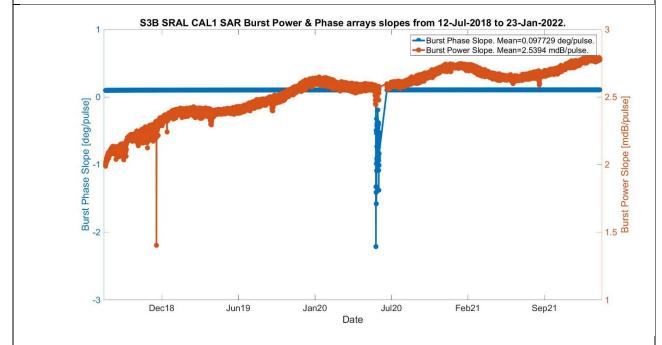


Figure 4-54. S3B CAL1 SAR Ku Phase & Power intra-burst corrections slopes along the whole mission.



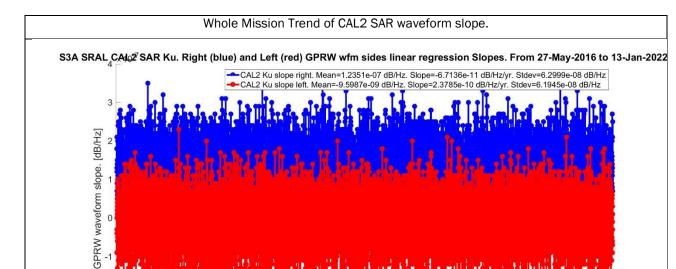


Figure 4-55. S3A Slope at each side of the CAL2 Ku waveform, averaged over the whole mission.

Mar19

Date

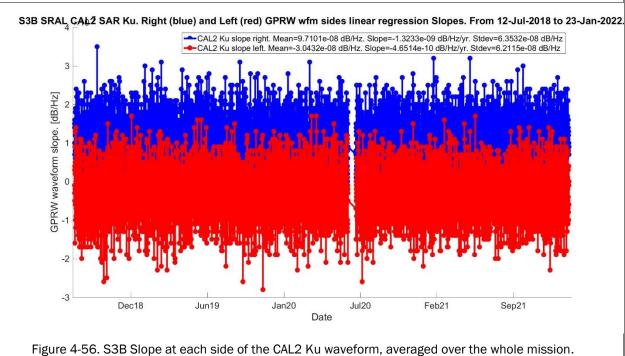
Jul20

Dec21

Nov17

-2

-3 Un16





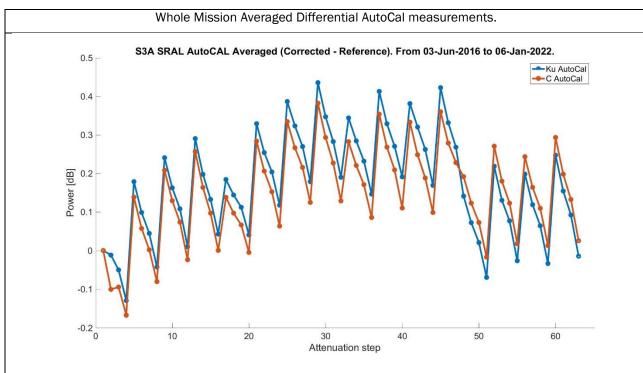


Figure 4-57. S3A Autocal measurements: Corrected - Reference. Averaged over the whole mission.

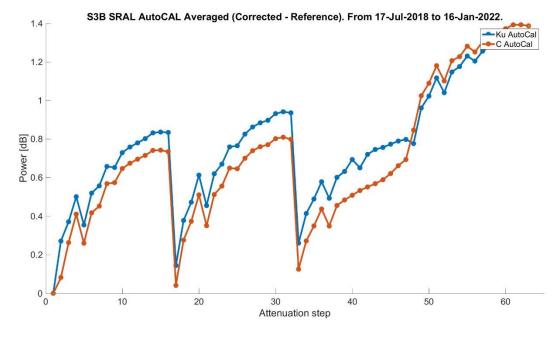


Figure 4-58. S3B Autocal measurements: Corrected - Reference. Averaged over the whole mission.



## AutoCAL Ku band attenuation progression series.

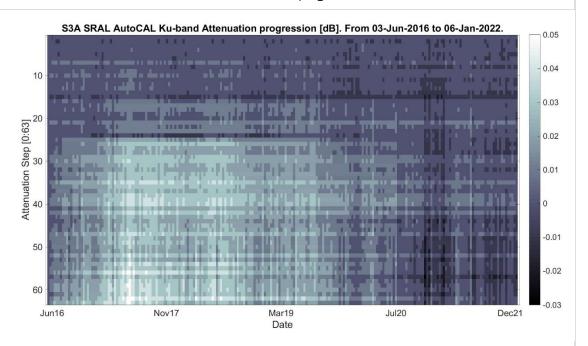


Figure 4-59. S3A AutoCAL Ku attenuation whole mission progression. Difference in dB with respect to the first attenuation value, for each attenuation step.

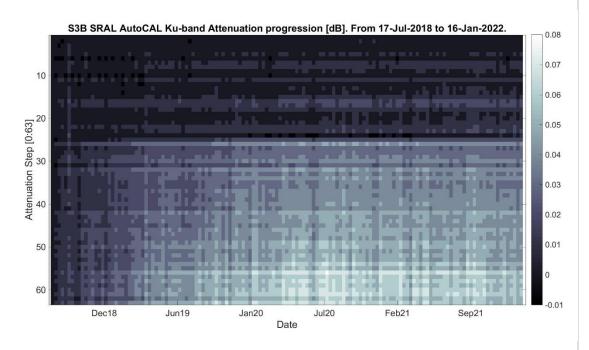
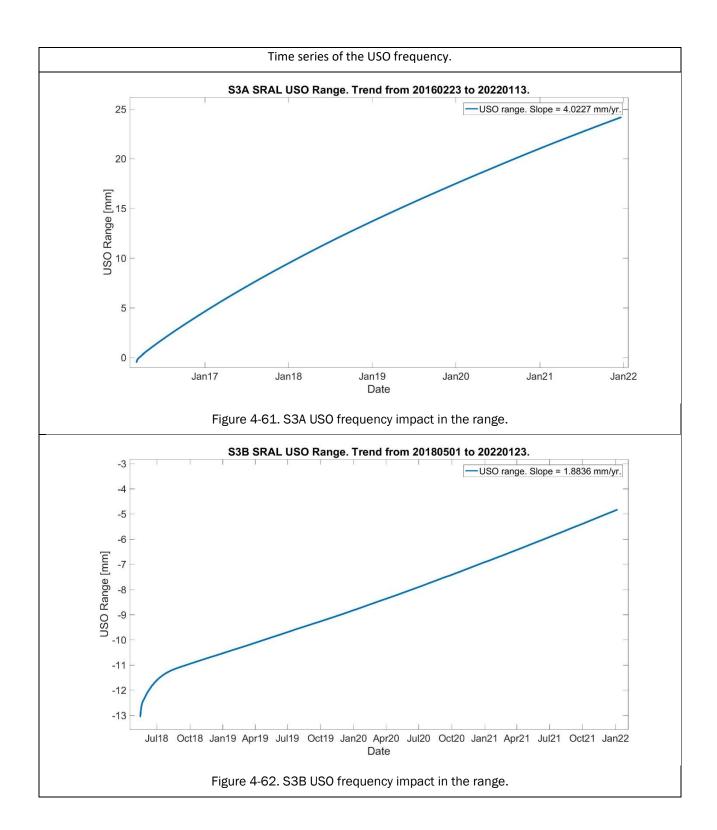
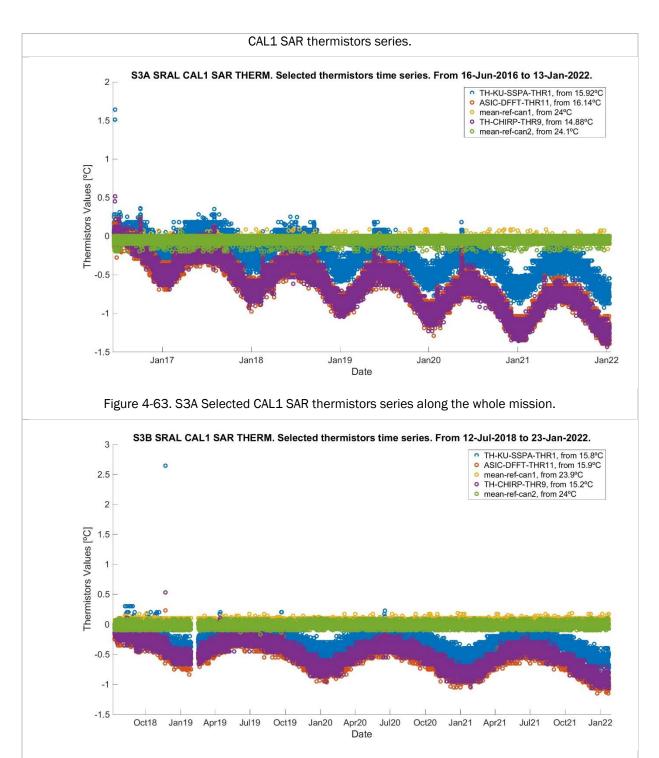


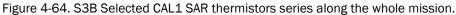
Figure 4-60. S3B AutoCAL Ku attenuation whole mission progression. Difference in dB with respect to the first attenuation value, for each attenuation step.













We can see from the above figures a general agreement between the S3A and S3B calibration parameters absolute values.

The most important and notable drift observed in the whole mission's series is the S3A CAL1 SAR Ku Power series, presenting a significant power decay at BOM (about -1 dB/year). Anyhow, we can see a slow stabilisation of this parameter along the mission, being below -0.5 dB/year from cycle 31.

A mission requirement of the CAL1 SAR Ku power is a maximum power drop of 3.5dB from beginning of mission. Below that power bound, the sea-ice and ice sheets geophysical measurements could be impacted, due to a poor SNR. A power model fitting has been computed for both S3A and S3B missions, and the assumption is to reach the limit by 2036.

For S3B, there are some spikes of CAL1 SAR Ku power of around 0.05 dB. These values coincide with products presenting a different calibration sequence, which could influence the operational point of the SRAL instrumentation, mainly SSPA/HPA. In addition, there are specific noisier periods during 2018, which are related to the existence of different operational modes (acquisition or tracking) before the calibration sequence activation, which impacts the initial conditions of the instrumental path performance. The rest of the periods are less noisy due to working in Open Loop tracking mode (where no acquisition mode is needed) except in cycle 40 where Close Loop was selected.

The S3A PTR time delay has also decreased its negative trend, being currently almost flat. The S3B time delay long term trend is positive, although it shows a flat period around end of 2018.

The Ku band PTR widths trends of both missions are similar to their standard deviation, and several orders of magnitude below their absolute values. The S3A mission presents a clearer negative long-term trend, while S3B changes locally its behaviour.

The intra-burst corrections series along the missions are quite stable, except for the S3B burst phase anomaly at cycle 40 (a wrap in the burst phase caused a huge jump in the series). The code fix for this anomaly is now implemented, and the data returned to its precedent nominal values. Anyhow this problem caused a stop of the calibration processing during the time it was fixed, and a gap in the S3B series will be present until a new reprocessing is completed from the new PB.

The S3A burst power slope shows a clear annual behaviour (oscillations of less than 0.2 mdB/pulse) due to sensibility to instrumental thermal changes. The annual behaviour is also noted in the S3B burst power figure, drawn over an added slope, plus a correlation with thermal events on-board (April & September 2019 and June 2020). The burst phase slope is increasing along the missions, around 1 mdeg/pulse per year for S3A and around 9 mdeg/pulse per year for S3B.

From the attenuation steps progression in dB we can check, for each ATT step, the delta in attenuation with respect to the first record value. The tendencies are visible for specific attenuations in each mission, with excursions (see colour code at right hand side) of less than 0.1 dB.

The CAL2 parameters behaviour is stable and nominal along the missions, with ripples of same magnitude and position in the CAL2 waveform.

The USO clock frequency impact in the range has trends around 4 mm per year for the S3A mission and around 2 mm per year for the S3B mission. S3A mission trend is slowly becoming stabler and for S3B in the recent cycles it is slightly increasing; for S3A from cycle 40 to cycle 78 the whole mission trend has flattened in 0.7 mm/year; for S3B from cycle 21 to cycle 38 it has flattened in 0.61 mm/year, and then raised in 0.04 mm/year since then to cycle 59. This behaviour is maintained in the last cycles.



The thermistors data series are showing annual oscillations, and a long-term cooling. At some dates there are increases of the S3A thermistors values of around 0.3°C, returning in a short term to its precedent values, with a limited impact in the calibration series.

There is a peak in the S3B temperatures on board at date 2018/11/22 (e.g., THR1 is 3°C up). This event affects slightly the calibration parameters (for instance the CAL1 SAR Ku Time Delay presents a jump up of 1mm). All monitored calibration parameters returned to its precedent values after the event.

In S3B, from cycle 21 - orbit 239 (2019/01/29) to cycle 22 - orbit 96 (2019/02/15), the thermistors values were not written in the TMs, and are consequently not represented in the figures. This anomaly was caused after a restart due to a SMUG event. A new command for a restart was executed and fixed the anomaly successfully.

The OLTC uploads have a limited but visible impact in some calibration parameters, mainly in the CAL1 power and the burst power. It is the case of S3A around 26/08/2021 and S3B around 19/08/2021. When an OLTC upload is carried out, the Close Loop tracking mode is activated, causing different initial conditions of the calibration sequence, depending on the operational tracking mode just before (tracking or acquisition).

The secondary lobes are monitored along the mission, representing the power distribution in the PTR, which is expected to be symmetrical around the main lobe. Its behaviour suffers variations when the operational (Closed Loop on-board tracking mode) and thermal (switch-off/on) conditions are not stable. This impact is limited, but not corrected for in the L1b processing (as the other CAL1 & CAL2 corrections). A numerical retracker, considering the PTR real shape, would be a good candidate to overcome this drawback.

For the two missions, the long-term drift for the SAR power variables is higher in absolute terms for the Ku band than for the C band, the Ku band ageing is faster than the one from C band, probably caused by the more stressed Ku band instrumental operations (e.g. bursts transmission & reception only in Ku band). All standard deviations in the figures are computed without detrending.

As a general observation, we can say that the behaviour of all calibration parameters is nominal.



## 5 Calibration with Transponder

One of the transponders (TRP) used for the Sentinel-3 calibration activity is located in Crete, Greece. It was developed at the Technical University of Crete for the ESA's Copernicus Earth Observation programme. This site has been named CDN1 Cal/Val and is located at 35.3379302808°N, 23.7795182869°E and 1048.8184 m height with respect to the WGS84 reference system.

isardSAT has processed the Crete TRP data from a list of L1A products. This processing method is explained in [Garcia-Mondejar et al. 2018]<sup>1</sup>. Passes with IPF-SR-1 Version 06.13 (cycle 3 to 23) use reprocessed L1A and L2 data provided on the ftp.s3rep.acri-cwa.fr FTP server. Passes from cycle 24 to 45 increase in IPF-SR-1 Version as newer ones become available, up to Version 06.14 for the most recent passes.

For S3A, the passes over Crete on cycles 13, 21, 54, 70 and 81 have not been analysed because the TRP was not switched on due to extreme climate conditions and passes on cycle 48 and 50 have not been analysed due to maintenance work. For S3B, cycles 1 to 7 and 15 to 18 have not been included as the satellite was not overflying the TRP. The Crete TRP was not switched on due to extreme climate conditions for S3B cycles 20 and 60.

Regarding the geophysical corrections, for the Crete measurements the ionospheric and wet/dry tropospheric corrections were extracted from the transponder auxiliary files provided by the MPC team. Then, the solid earth, geocentric tide and ocean loading corrections are selected from the L2 products.

Since the month of June 2019, the Svalbard transponder has also been used to calibrate the Sentinel-3 satellites. The transponder is located at 78.23052306°N, 15.39376997°E and 492.772 m of height with respect to the WGS84 reference system.

For the Svalbard transponder, the ionospheric and wet/dry tropospheric, the solid earth, geocentric tide and ocean loading corrections are selected from the L2 products.

The range and datation results detailed here below are extracted from the minimisation of the RMS between theoretical and measured series. The range bias is computed as measured minus theoretical.

Table 3-1 presents, for each mission, the range and datation processing results from the Crete TRP latest cycle, together with a whole mission average and standard deviation. The results for S3A show a negative measured range, 9.67 mm smaller than expected (elevation 9.67 mm higher than expected), and a datation bias of -117.33 microseconds. They also show a 0.75 mm stack noise. For S3B, the results show a negative measured range, 40.45 mm smaller than expected (elevation 40.45 mm higher than expected), and a datation bias of -22.21 microseconds. They also show a 0.73 mm stack noise. It is interesting to note that for S3B the results from cycle 21 show that the datation bias has been reduced from -114.60 microseconds (passes were in tandem orbit following S3A) to residual values.

Table 3-2 presents the same kind of results, from the TRP passes over Svalbard. For S3A, the results show a negative measured range, 67.53 mm smaller than expected (elevation 67.53 mm higher than expected), and a datation bias of -74.90 microseconds. They also show a 17.26 mm stack noise. The results for S3B show a negative bias range, 78.18 mm smaller range than expected (elevation 78.18 mm higher than expected), and a datation bias of -37.14 microseconds. They also show a 9.57 mm stack noise (we have less than 1 mm with S3A/B over Crete and 7.4 mm with CryoSat-2 over Svalbard)

<sup>&</sup>lt;sup>1</sup> Garcia-Mondéjar, Albert, et al. "CryoSat-2 range, datation and interferometer calibration with Svalbard transponder." Advances in Space Research 62.6 (2018): 1589-1609.



**Figure 5-1** to **Figure 5-4** depicts the series of TRP processing results for the two missions and the two transponders, including range, datation, stack alignment and stack range noise.

During the Commissioning Phase of S6, the TRP located in Crete was rechecked, and the internal delay value was modified, with this new configuration, the range biases have been recalculated for all passes (S3A/B).

On the other hand, at the last CM in November 2021, S. Dinardo (CLS) showed that the IPF was reading the USO delta frequency from the USO ASCII files overlooking the sign. This affects the S3B mission, which present negative values along the full series. This impact has been accounted for. The TRP processing results have been updated accordingly. Before applying this correction, the S3B range drift was of – 5 mm/year, while currently the regression line for S3B shows a drift of - 0.51 mm/year. This drift is well aligned between all the passes over Crete and Svalbard.

The last passes over Svalbard, both S3A and S3B are having strong interferences, making it very difficult to retrieve a clear signal from the TRP. As shown in the tables and figures, the stack range noise is quite different from their usual values. There seems to be a similar effect in general in the last summer periods, with an increasing stack range noise. This issue will be monitored in the following passes.

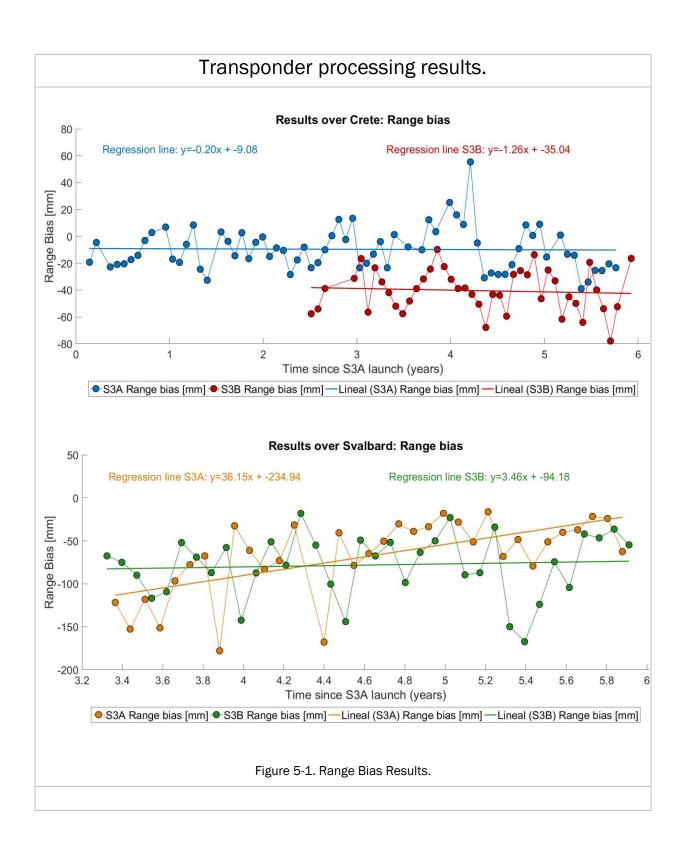
Cycle – Mission	Date	Range bias [mm]	Datation bias [microseconds]	Alignment [mm/beam]	Noise [mm]	IPF-SR-1 Version
80 - S3A	2021/12/18	-	-	-	-	06.14
61 - S3B	2022/01/20	-16.46	-25.47	0.04	0.50	06.14
Mean S3A (70 passes)		-9.67	-117.33	0.06	0.75	-
Standard Deviation S3A		15.87	20.09	0.01	0.18	-
Mean S3B (43 passes)		-40.45	-22.21	0.02	0.73	-
Standard Deviation S3B		15.85	20.44	0.01	0.16	-

Table 5-1. Results of Crete TRP passes processing

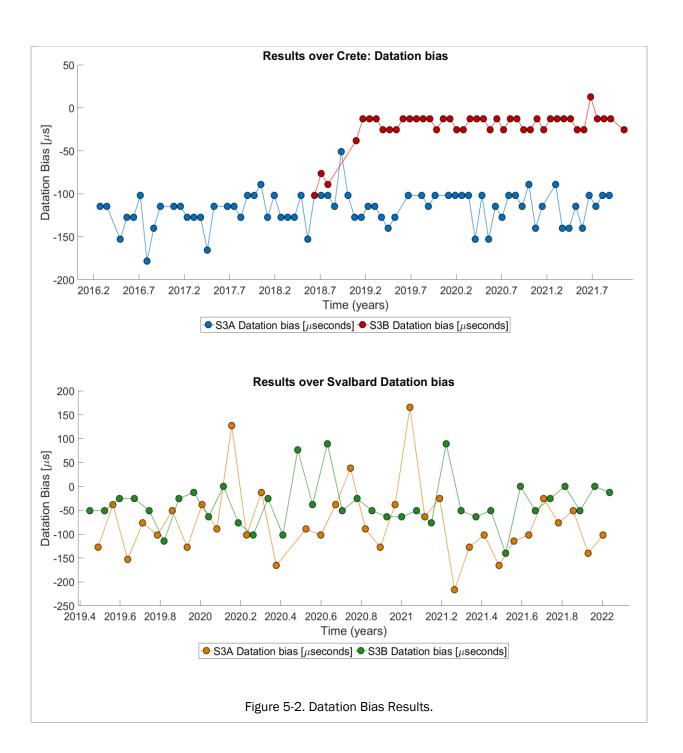
Cycle – Mission	Date	Range bias [mm]	Datation bias [microseconds]	Alignment [mm/beam]	Noise [mm]	IPF-SR-1 Version
80 - S3A	2022/01/02	-62.54	-74.90	0.08	10.00	06.14
61 - S3B	2022/01/14	-54.71	-12.74	0.03	7.88	06.14
Mean S3A (34 passes)		-67.53	-74.90	0.07	17.26	-
Standard Deviation S3A		43.91	76.13	0.04	11.37	-
Mean S3B (36 passes)		-78.18	-37.14	0.02	9.57	-
Standard Deviation S3B		36.51	49.31	0.04	8.63	-

Table 5-2. Results of Svalbard TRP passes processing

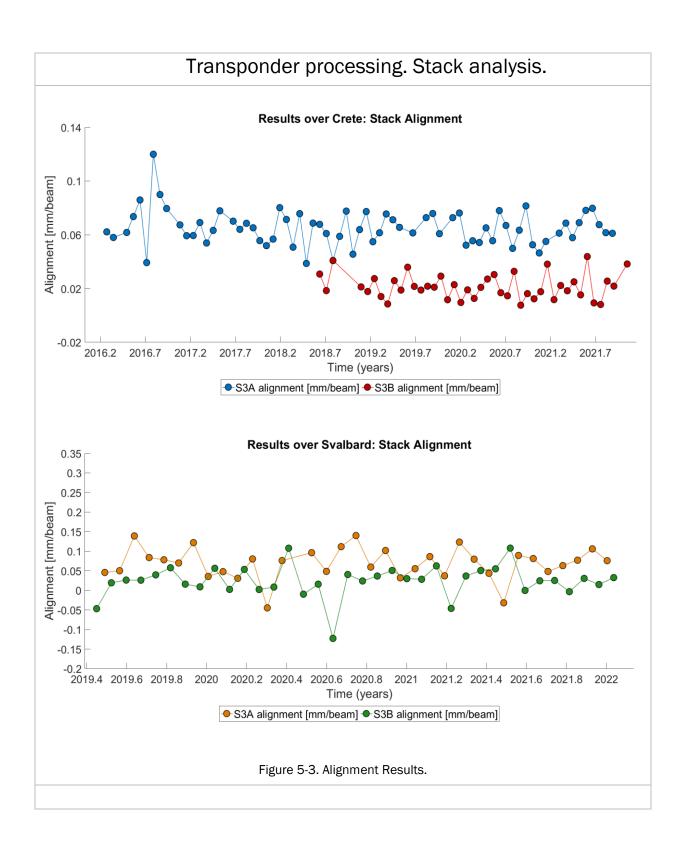




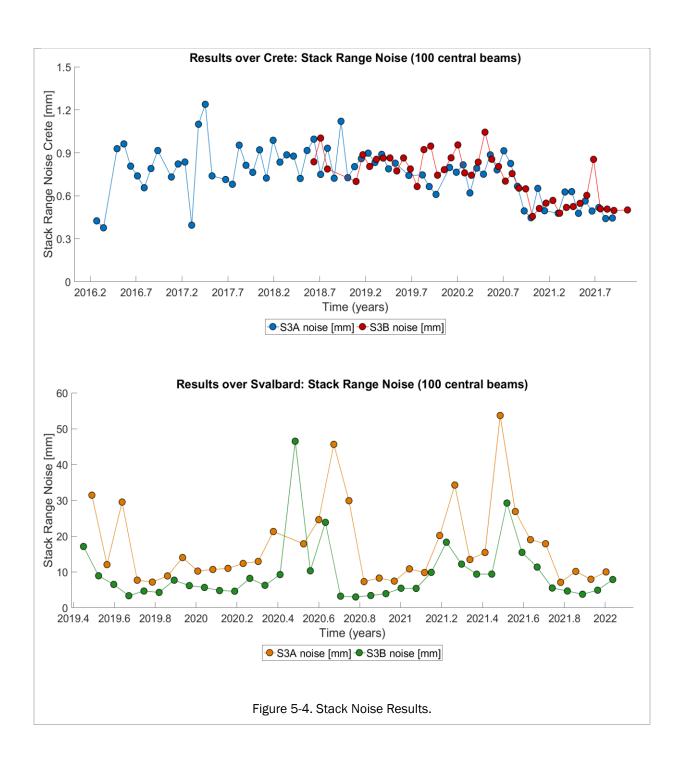














## Appendix

The Product Format Specification applicable to the S3 STM CAL Level-1B products assessed in this report is available in Sentinel Online, version 2.13:

 $\frac{https://sentinel.esa.int/documents/247904/0/Sentinel-3-Product-Data-Format-Specification-Level-1-products/2b7c773b-44ca-447e-9b86-f7ebd231261c}{$ 

